

Master Checklist

(PA-28-161)

(PA-28-181)

(PA-44-180)

Introduction

This document has been prepared by North Star Aviation, Inc. and is being furnished to students of the flight school in accordance with 14 CFR Part 141.75(a). To better serve our students and promote future improvements to be made, the formatting of this document has changed. Previous versions where provided in separate books for each aircraft. This book contains the required normal and emergency procedures for the safe operation of our aircraft as specified in the respective Piper Information Manuals. In addition to procedures required by Piper, some NSA specific procedures have been added for reference.

Organization

This document has been divided into three separate sections: PA-28-161/181, PA-44 -180, and Appendices. The two aircraft sections are further divided into Normal and Emergency procedures.

Inside the PA-28 section, procedures specific to the Archer (181) will be found in GOLD.

Inside the PA-44 section, procedures specific to non-G1000 equipped aircraft will be found in PURPLE.

In each section, procedures surrounded in red dotted boxes are considered "Memory Items". These procedures are required to be memorized and will be evaluated periodically throughout training.

Revision Control

To allow for improvements based on safety, performance, or standardization, this document is being provided in an easily revisable format. It is the student's responsibility to ensure their copy of this document contains the most current revision(s). A record of revisions can be found on the next page that will track individual page numbers affected by each revision. Each individual page will have its revision number printed in the lower left-hand corner.

Disclaimers

This document is provided for training purposes only and is not intended to replace the required Pilot Operations Manual in each aircraft. In actual Emergency Situations, the POH should be consulted if able. No replication of this document is authorized without the express written consent of North Star Aviation, Inc.

Record of Revisions				
<u>Number</u>	<u>Date</u>	Summary of Changes	Affected Pages	
Original	July 13, 2011	Original Issuance of Warrior checklist	ALL	
Original	Aug. 01, 2011	Original Issuance of Seminole checklist	ALL	
Rev 1	Nov. 16, 2011	First Rev. to Warrior Checklist	3, 4, 9, 10, 11, 36	
Rev 1	Feb. 22, 2012	First Rev. to Seminole Checklist	: ALL	
Rev 2	Aug. 02, 2012	Second Rev. to Warrior and Seminole Checklists	3, 4, 9, 10, 11, 36 3, 12, 19, 32	
Rev 3	July 01, 2014	Third Rev. to Warrior Checklist	3, 14-36	
Rev 3/4	Jan. 01, 2015	Third Rev. to Seminole Checklis Fourth Rev. to Warrior Checklis		
Rev 4	Jan. 13, 2015	Fourth Rev. to Seminole Check	list 9, 10, 17	
Rev 5	April 13, 2015	Fifth Rev. to Warrior Checklist	2, 9, 12, 18, 19	
Rev 5	Feb. 22, 2016	Addition of G1000 procedures the Seminole checklist	to ALL	
Rev 6	Jan. 13, 2016	Sixth Rev. to Warrior Checklist	9	
Rev 7	Aug. 27, 2018	Combination of Warrior, Arche Seminole, and company Procedures in one manual.	r, ALL	
Rev 8	Jan. 8, 2019	Small Formatting changes, Updated Procedures for G1000 NXi and Fuel Injected Engines.	ALL	

Reco	rd of	Revisions	: Cont'd
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Number Date	Summary of Changes	Affected Pages

<u>NOTE</u>			CHECK
Prior to beginning the preflight wa	· · · · · · · · · · · · · · · · · · ·		CLOSE AND SECURE
fuel, and windshield and notify lir	ne service if needed.		CHECK
COCKPIT INSPECTION			CLEAR
AIRCRAFT DOCUMENTS	VERIFY		CHECK TENSION
HOBBS/TACHTIME	CHECK		CHECK
PARKING BRAKE	SET		APPROX. 3.25" INFLATION
COMPASS DEV CARD	CHECK		CHECK
PITOT/STATIC	DRAIN	CHOCKS	REMOVE
MAGNETO SWITCH(ES)	OFF		
ELECTRICAL SWITCHES	OFF	NOSE SECTION LEFT SIDE	='
AVIONICS	OFF		CHECK
CIRCUIT BREAKERS	CHECK IN	ENGINE BAFFLE SEALS	CHECK
BATTERY/MASTER SWITCH	ON		SECURE
FLIGHT CONTROLS		COWLING	CLOSE AND SECURE
FUEL GUAGES	CHECK	GASCOLATOR	DRAIN
ANNUNCIATOR PANEL LIGHTS	CHECK	WINDSHIELD	CHECK
EXTERNAL LIGHTS	ON/CHECK	TEMP PROBE	CHECK
PITOT HEAT			
STALL WARNING HORN		LEFT WING	
EXTERNAL LIGHTS	OFF	FRESH AIR INLET	CLEAR
PITOT HEAT		MAIN GEAR STRUT	APROX 4.5"
BATTERY/MASTER		TIRE/BRAKEPAD/DISC	CHECK
EMPTY SEAT BELTS		TIE DOWN/CHOCK	REMOVE
STAB/RUDDER TRIM		FUEL TANK SUMP	DRAIN
FLAPS		FUEL VENT	CLEAR
WINDOWS/WINDSHIELD		FUEL TANK	CHECK QUANTITY
TOW BAR/BAGGAGE		LEADING EDGE	CHECK
		PITOT MAST	UNOBSTRUCTED
PREFLIGHT INSPECTION		TEMP PROBE	CHECK
		STALL WARNING	CHECK
RIGHT WING		WING TIP AND LIGHTS	CHECK
FLAPS/HINGES	CHECK	WING SURFACES	CHECK
AIRLERON AND HINGES		STATIC WICKS	CHECK
STATIC WICKS		AILERON AND HINDGES	CHECK
WING SURFACES		FLAPS AND HINGES	CHECK
WING TIPS AND LIGHTS			
LEADING EDGE	CHECK	FUSELAGE LEFT SIDE	
FUEL TANK		GENERAL CONDITION	CHECK
FUEL VENT	•	ANTENNAS	CHECK
FUEL TANK SUMP		UNDERBELLY	CHECK
TIE DOWN/CHOCK			
MAIN GEAR STRUT		EMPENNAGE	
TIRE/BRAKEPAD/DISC		GENERAL CONDITION	CHECK
FRESH AIR INLET		FRESH AIR INLET	CHECK
DOOR HINGES		RUDDER	CHECK
WINDSHIELD		STABILATOR	CHECK
		ANTISERVO TAB	CHECK
NOSE SECTION RIGHT SIDE		TIE DOWN	REMOVE AND CHECK
ENGINE COMPARTMENT	CHECK		
ENGINE BAFFLE SEALS		FUESLAGE RIGHT SIDE	
OIL		GENERAL CONDITION	CHECK
		ANTENNAS	CHECK

UNDERBELLY......CHECK

START UP CHECKLIST	ENGINE START FLOODED
DEFORE STARTING FAIGHE	BATTERY/MASTER SWITCH ON
BEFORE STARTING ENGINE	*PFDVERIFY CORRECT SOFTWARE
FLAPS RETRACT	ALTERNATORON
PARKING BRAKE SET	**STANDBY ALTERNATORON
PASSENGER BRIEFINGCOMPLETE	ELETRIC FUEL PUMPOFF
SEAT BELTSFASTEN	STROBE LIGHTSON
CIRCUIT BREAKERSCHECK IN	MIXTUREIDLE CUTOFF
FUEL SELECTORDESIRED TANK	THROTTLEOPEN FULL
AVIONICS/ELECTR SWITCHES OFF	PROP AREA CLEAR
CARB HEAT /ALT AIROFF	STARTERENGAGE
E-BATT ARM	MIXTURE FULL RICH
*E VOLTSVERIFY 23.3	THROTTLE 1000RPM
FUEL QUANTITY CHECK	OIL PRESSURE CHECK
*If E-volts are less than 23.3, the voltage can be	ANNUCIATOR PANELLGTS OUT
checked again at the end of the GROUND CHECK. E-	AMMETERCHECK
volts must not be less than 23.3 prior to flight.	
	*ONLY IN GLASS AIRCRAFT
NORMAL START	**ONLY IF STANDBY ALTERNATOR INSTALLED
BATTERY/MASTER SWITCHON	
PFDVERIFY CORRECT SOFTWARE	BEFORE TAXI CHECK
ALTERNATOR SWITCHON	BATT MASTR SWITCHVERIFY ON
**STANDBY ALTERNATORON	AVION MASTER SWITCHON
ELECTRIC FUEL PUMPON	CIRCUIT BREAKERSCHECK IN
STROBE LIGHTSON	MFDCHECK DATABASE CURRENCY
PROP AREACLEAR	FUEL TOTALIZER SYNC OR MANUAL
CARBURETED	FUEL SELECTORSWITCH TANKS
MIXTURERICH	
	CAS MESSAGES
THROTTLE	PFD ANNUNCIATORSCHECK
PRIMER AS REQUIRED	MSG SOFTKEYCHECK
STARTERENGAGED	STANDBY INSTRUMENTSCHECK
FUEL INJECTED — COLD	LIGHTS AS REQUIRED
LEFT/RIGHT MAGNETOSON	CABIN HEAT AS REQUIRED
THROTTLE 0.25" OPEN	ATIS/AWOS OBTAIN
MIXTURE PRIME THEN CUT OFF	GPS/NAV/RADIOSSET
STARTERENGAGED	ALTIMETERSET
MIXTUREADVANCE	STANDBY ALTSET
FUEL INJECTED — HOT	TRANSPONDERALT
LEFT/RIGHT MAGNETOSON	HEADING INDICATORSET
THROTTLE 0.5" OPEN	MFD SETAIRPORT DIAGRAM
MIXTURECUT OFF	TAXI BRIEFCOMPLETE
STARTERENGAGED	PARKING BRAKE RELEASE
MIXTUREADVANCE	
THROTTLE 1000RPM	TAXI CHECK
	PARKING BRAKERELEASE
DIL PRESSURE CHECK	
	TAXLAREA CLEAR
ELECTRIC FUEL PUMP OFF	
DIL PRESSURE CHECK ELECTRIC FUEL PUMP OFF FUEL PRESSURE CHECK ANNUCIATOR PANEL LGTS OUT	THROTTLEADVANCE SLOWLY
ELECTRIC FUEL PUMP OFF	TAXI AREA
ELECTRIC FUEL PUMP OFF EUEL PRESSURE CHECK ANNUCIATOR PANEL LGTS OUT	THROTTLEADVANCE SLOWLY BRAKES/STEERINGADVANCE SLOWLY

ENGINE SHUTDOWN

RUN UP/GROUND CHECK	BEFORE TAKE OFF CHECK	
BRAKES SET/HOLD	FLIGHT INSTRUMENTS	RE-CHECK/SET
FLIGHT CONTROLS FREE &CORRECT	ELECTRIC FUEL PUMP	ON
MIXUTREFULL RICH	FUEL SELECTOR	FULLEST TANK
THROTTLE2000 RPM	LANDING LIGHT	ON
ENGINE GAUGESIN LIMITS	RECOG LIGHT	ON
MAGNETOS MAX DROP 175 RPM	PITOT HEAT	. AS REQUIRED
MAX DIFFERENCE 50 RPM	ENGINE GAUGES	-
CARB HEAT/ALT AIRCHECK(~75 RPM)	CARB HEAT/ALT AIR	OFF
*VACUUM4.8" - 5.2"	FLAPS	
*IF EQUIPPED WITH VACUUM	TRIM	SET
	MIXTURE	
IF E-VOLTS WERE LESS THAN 23.3:	TRANSPONDERCO	
E-VOLT CHECK	TIME OFF	•
BATT MASTEROFF	TAKE OFF CLEARANCE	
ALTERNATOR SWITCH OFF	FINAL APPROACH AREA	
E-VOLTSCHECK	THAL AT HOACH AREA	CLLAN
BATT/ALTERNATOR SWITCHESON	CLIMB CHECK—ABOVE 1000' AGL	
	FLAPSVERI	EV RETRACTED
STANDBY ALTERNATOR CHECK(IF EQUIPPED)	ELECTRIC FUEL PUMP	
ALTERNATOROFF	CRUISE CLIMB AIRSPEED	
ALT INOP ANNUNCIATOR VERIFY ON	DEPARTURE AREA	
STBY ALT ANNUNCIATOR VERIFY ON	DEFARTORE AREA	CLLAN
THROTTLE1000 RPM	CRUISE CHECK	
ALTERNATOR ON	CRUISE POWER	CET
STBY ALT ANNUN VERIFY OUT	MIXTURE	
STANDBY INSTCHECK	LANDING LIGHT	
	RECOG LIGHT	
RUN UP/GROUND CHECK CON'TD	HEADING INDICATOR	
ANNUN. LIGHTSPRESS TO TEST	VERIFY FUEL PUMP	
THROTTLEIDLE	VERIFT FUEL PUIVIP	UFF
RPM/OIL PRESSURECHECK		
**AUX VACUUMCHECK		
THROTTLE1000 RPM		
ALTERNATE STATICCHECK		
TAKE OFF BREIFINGCOMPLETE		
DOOR/WINDOWCLOSED		
IFR CLEARANCE(IF REQUIRED)ACQUIRE		
**IF EQUIPPED WITH AUX VACUUM		
PRE-MANUEVER CHECK MIXTURESET	POST-MANUEVER CHECK LANDING LIGHT	055
ELECTRIC FUEL PUMPON	RECOG LIGHT	
ELECTRIC FUEL PUIVIPON		
FUEL FOR TAXABLE	FUEL	FULLEST TANK
LANDING LIGHT ON	ELECTRIC FUEL PUMP	OFF
LANDING LIGHT ON RECOG LIGHT ON		OFF
LANDING LIGHT ON RECOG LIGHT ON MINIMUM SAFE ALT VERIFY	ELECTRIC FUEL PUMP	OFF
FUEL	ELECTRIC FUEL PUMP	OFF

IN RANGE CHECK—15NM OR LESS	
WEATHER OBTAIN	
INSTRUMENTS SET	
RADIOS SET	
ENVIRONMENTBRIEF	
ENVINORIVIENT	
DESCENT CHECK—10NM OR LESS	
SEATBELTS SECURE	
FUEL SELECTORFULLEST TANK	
MIXTURERICH	
LANDING LIGHTSON	
RECOG LIGHTSON	
DESCENT POWER AS REQUIRED	
BEFORE LANDING CHECK	
ELECTRIC FUEL PUMPON	
FUELVERIFY FULLEST TANK	
MIXTURERICH	
SEATBELTS FASTENED	
CARB HEAT/ALT AIR AS REQUIRED	
AFTER LANDING CHECK	
RADIO CALLREPORT CLEAR	
FLAPS RETRACT	
CARB HEAT/ALT AIROFF	
ELECTRIC FUEL PUMPOFF	
LANDING LIGHT AS REQUIRED	
RECOG LIGHT AS REQUIRED	
PITOT HEATOFF	
TRANSPONDERALT/1200	
FLIGHT PLANCLOSED	
TAXI CLEARANCEOBTAIN	

ENGINE SHUTDOWN	
BRAKES	SET/HOLD
THROTTLE	1000 RPM
AVION MASTER	OFF
LANDING LIGHT	OFF
RECOG LIGHT	OFF
NAV LIGHT	OFF
ELECTRICAL EQUIPMENT	OFF
EMER BATT	OFF
*THROTTLE	IDLE
*MAGNETOS	GROUND CHECK
*THROTTLE	1000 RPM
MIXTURE	IDLE CUT-OFF
THROTTLE	IDLE
MAGNETOS	OFF
STROBE LIGHTS	OFF
ALTERNATOR	OFF
STBY ALTERNATOR	OFF
BATTERY/MASTER	OFF
STANDBY INSTRUMENT	
PARKING BRAKE	RELEASE
*ONLY IN WARRIOR	
SECURING AIRCRAFT	
HOBBS/TACH	RECORD
CABIN INTERIOR	CLEAN
MIXTURE	
MASTER	OFF
MAGS	OFF
DOOR/WINDOW	CLOSE
CHOCK	
TIEDOWNS	
OIL HEATER	PLUG IN
BLANKET	ON

POST-FLIGHT INSPECTIONCOMPLETE

EMEREGENCY PROCEDUI	RES	ENGINE POWER LOSS DURING FLIGHT
		<u>A</u> IRSPEEDV _G
ENGINE FIRE DURING START		LANDING AREA BEST SUITABLE
STARTER	CRANK ENGINE	R ESTARTATTEMPT
MIXTURE		FUELSWITCH TO TANK WITH FUEL
THROTTLE		ELECTRIC FUEL PUMPON
ELECTRIC FUEL PUMP		MIXTURE RICH
FUEL SELECTOR		CARB HEAT/ALT AIRON
ABONDON AIRCRAFT IF FIRE CO		ENGINE GAUGES CHECK
EXTERNAL FIRE EXTINGUISHER.		THROTTLECHECK FOR POWER
ENGINE POWER LOSS DURING	TAKEOFF	If power has not been restored:
If sufficient runway remains for		IGNITION SWITCHL, R, THEN BOTH
land straight ahead.		THROTTLE & MIXTURETRY DIFFERENT
		<u>M</u> AYDAY121.5/SQ 7700
If insufficient runway remains,	maintain a safe air-	S ECUREPerform Power Off Landing Checklist
speed. Make only shallow turn		
and land. Set flaps as the situat		If power is restored:
	- 11 21	CARB HEATOFF
f sufficient altitude has beer	gained to attempt a	ELECTRIC FUEL PUMPOFF
estart:		
AIRSPEED	PITCH FOR V _G	POWER OFF LANDING
FUEL SELECTOR	SWITCH TANK	When committed to landing:
ELECTRIC FUEL PUMP		MAGNETOSOFF
MIXTURE	VERIFY FULL RICH	FUEL SELECTOROFF
CARB HEAT/ALT AIR		BATTERY/MASTEROFF
f power is not restored, procee	ed with power off	ALTERNATOROFF
landing.		STBY ALTERNATOROFF
		ELECTRIC FUEL PUMPOFF
ENGINE FIRE IN FLIGHT		MIXTUREIDLE CUT-OFF
FUEL SELECTOR	OFF	THROTTLECLOSE
THROTTLE	CLOSED	SEATBELTSTIGHT
MIXTURE	IDLE CUT-OFF	CABIN DOORUNLATCH/OPEN
ELECTRIC FUEL PUMP	OFF	
CABIN HEAT	OFF	Landing:
DEFOSTER	OFF	AIRSPEED
IF FIRE PERSISTS		FLAPS FULL
EMERGENCY DESCENT	INITIATE	
Proceed with power off landing	g.	
		STARTER ENGAGED
ELECTRICAL FIRE/SMOKE IN FLI	<u>GHT</u>	ON GROUND
EMER BATT		THROTTLE REDUCE
BATTERY MASTER	OFF	ENG START CIRCUIT BREAKER PULL
ALTERNATOR	OFF	ENGINE SHUTDOWN
*STANDBY ALTERNATOR	OFF	IN FLIGHT
*STANDBY ATTITUDE	VERIFY ON	THROTTLEREDUCE
VENTS	OPEN	ENG START CIRCUIT BREAKER PULL
CABIN HEAT	OFF	Land as soon as possible.
FIRE EXTINGUISHER	IF NEEDED	
IF FIRE PERSISTS	•	
EN 4ED GENIOV DESCE:-		

FAILURE OF DED	FAILURE OF AIR DATA ATTITUDE AND HEADING
FAILURE OF PFD	FAILURE OF AIR DATA, ATTITUDE, AND HEADING
INDICATION: PFD GOES BLANK	REFERENCE SYSTEM
STANDBY INSTRUMENTVERIFY ON(NO FLAG)	INDICATION: AIRSPEED, ATTTUDE, HEADING AND ALTITUDE REPLACED WITH RED X'S
IF TIME PERMITES:	STANDBY INSTRUMENT VERIFY ON (NO FLAG)
PFD BRIGHTNESS CONTROL ADJUST TO FULL	STANDBY INSTRUVIENT VERIFY ON (NO FLAG)
PFD CIRCUIT BREAKERPULL AND RESET	IF TIME PERMITS:
REVERSIONARY MODE (IF EQUIPPED)ACTIVATE	NON G1000
REVERSIONART WIODE (II EQUIFFED)ACTIVATE	PFD CIRCUIT BREAKER PULL AND RESET
IF PFD SCREEN CANNOT BE RESTORED:	G1000 EQUIPPED
SECONDARY CDI SOURCEULTILIZE	ADAHRS BREAKERPULL AND RESET
ENGINE INSTRUMENTS REFER TO MFD	ADATING BREAKEN TOLE AND RESET
ENGINE INSTROMENTS REFER TO WILD	IF ADAHRS INITIALIZATION DOES NOT OCCUR:
NOTE	SECONDARY CDI SOURCEUTILIZE
Maintain attitude, airspeed, and heading control using	ENGINE INSTRUMENTSREFER TO MFD
standby instruments, magnetic compass and other	ENGINE INSTRUMENTSREI ER TO WILD
- · · · · · · · · · · · · · · · · · · ·	NOTE
directional indicators. (GPS NAV page group).	
LAND AS COON AS DRACTICAL	Maintain attitude, airspeed, and heading control using
LAND AS SOON AS PRACTICAL	standby instruments, magnetic compass and other
LOCC OF DED ENGINE DATA	directional indicators. (GPS NAV page group).
ENGINE INSTRUMENTS REFER TO MFD	LAND AS SOON AS PRACTICAL
ENGINE INSTRUMENTS REFER TO MIFD	LAND AS SOON AS PRACTICAL
LAND AS SOON AS PRACTICAL	INVALID ATTITUDE AND HEADING DATA
LAND AS SOON AS PRACTICAL	INDICATION: ATTITUDE AND HEADING DATA INDICATION: ATTITUDE AND HEADING DATA RE-
LOSS OF MED (C1000)	MOVED AND REPLACED WITH RED X'S
LOSS OF MFD (G1000)	
REVERSIONARY MODEACTIVATE	STANDBY INSTRUMENT VERIFY ON (NO FLAG)
EVIT /AV/OID IED I AND AS SOON AS DRACTICAL	Maintain attitude control using standby attitude indicator.
EXIT/AVOID IFR. LAND AS SOON AS PRACTICAL	cator.
INVALID AIR DATA	IF TIME PERMITS:
INDICATION: AIRSPEED/ALTIMETER/VSI REPLACED	PFD CIRCUIT BREAKERPULL AND RESET
WITH RED Xs	Maintain attitude, airspeed, and heading control using
WITH RED AS	standby instruments, magnetic compass and other
Maintain aircraft control by referring to standby in-	directional indicators. (GPS NAV page group).
struments	LAND AS SOON AS PRACTICAL
Struments	LAND AS SOON AS FILACTICAL
PFD CIRCUIT BREAKERPULL AND RESET	ELECTRIC TRIM RUNAWAY
IF AIR DATA STILL INVALD:	CONTROL WHEEL GRASP FIRMLY
REFER TO STANBY INSTRUMENTS	ATTITUDE INDICATORCROSS CHECK
RELEKTO STANDT INSTRUMENTS	TRIM INTERUPTPRESS AND HOLD
LAND AS SOON AS PRACTICAL	PITCH TRIM CIRCUIT BREAKERPULL
LAND AS SOON AS I NACTICAL	PITCH TRIM RETRIM MANUALLY
INVALID HEADING DATA	THEIT INIVI
INDICATION: HSI REPLACED WITH RED Xs	COMM 1 AND 2 FAILURE
INDICATION: HIS REFEREED WITH RED AS	AUDIO MKR CIRCUIT BREAKERPULL
Maintain aircraft control by referring to standby in-	ACOTO MICH CINCOTI DILFACETAMINATOLE
struments. Maintain heading using secondary heading	NOTE
source such as magnetic compass or GPS NAV page	If power is lost to the audio panel a fail-safe communi-
group	cations path becomes available between the pilot's
_θ ι σα ρ	headset/microphone and COM1.
	nedusely interophone and colvis.

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*IF EQUIPPED

EXIT/AVOID IFR AS SOON AS PRACTICAL

EMERGENCY DESCENT.....INITIATE

LAND AS SOON AS PRACTICAL

PFD CIRCUIT BREAKERPULL AND RESET

LOSS OF HEADING ACCURACY

INDICATIONS: DIFFICULTY MAINTAINING CDI COURSE, Monitor engine temperature gauges. **EXCESSIVE DIFFERENCE BETWEEN HEADING AND** TRACK, ATC INDICATES THE AIRCRAFT IS ON THE WRONG HEADING, EXCESSIVE DEVIATION BETWEEN PFD AND MAG COMPASS.

IF HEADING DIFFERS MORE THAN 10° FROM COMPASS, USE THE MAG COMPASS FOR PRIMARY HEADING IN-FORMATION.

TOTAL LOSS OF ENGINE INSTRUMENTS

NON G1000

DAU CIRCUIT BREAKERPULL AND RESET **G1000 EQUIPPED**

GEA CIRCUIT BREAKERPULL AND RESET IF SPECIFIC ENGINE DATA IS STILL INVALID:

NOTE

THE FOLLOWING ENGINE MESSAGE(S) WILL BE DIS-PLAYED ON THE MFD FOR THE INVALID PARAMETER

- CHECK OIL TEMP
- CHECK OIL PRESS
- **CHECK RPM**
- CHECK CHT
- **CHECK EGT**

IF FAILURE OCCURS DURING TAKEOFF:

RETURN FOR LANDING	
THROTTLEF	ULL
MIXTURE	ICH

IF FAILURE OCCURS DURING CLIMB OR LANDING:

MIXTURE	RICH
THROTTLE	AS REQUIRED

IF FAILURE OCCURS AFTER SETTING CRUISE POWER AND MIXTURE:

POWER MAINTAIN POWER SETTING LAND AS SOON AS PRACTICAL

IF FAILURE OCCURS PRIOR TO OR DURING DESCENT: MIXTURE.....RICH THROTTLE..... SET FOR 500 FPM PITCH FOR A/S

..... DESCEND AT DESIRED A/S OR 126 KTS

LAND AS SOON AS PRACTICAL

LAND AS SOON AS PRACTICAL

CROSS CHECK MONITOR

YELLOW CROSSCHECK ATTITUDE ANNUN. ON PFD. STRAIGHT, LEVEL, UNACCEL FLIGHT ESTABLISH AIRCRAFT ATTITUDE REFERENCE STANDBY LOSS OF OIL PRESSURE

Land as soon as practical and investigate cause. Prepare for power off landing.

HIGH OIL TEMPERATURE

MIXTURERICH	
THROTTLE REDUCE	
PITCH REDUCE TO INCREASE AIRLOW	
Land at nearest airport and investigate the problem	١.
Prepare for power off landing.	

LOSS OF FUEL PRESSURE/FLOW

ELECTRIC FUEL PUMPON FUEL.....SWITCH TO TANK WITH FUEL Land as soon as practical.

ENGINE ROUGHNESS

CARB HEAT/ALT AIR	ON
If roughness continues after	er one minute:
CARB HEAT	OFF
MIXUREAD	JUST FOR SMOOTHNESS
ELETRIC FUEL PUMP	ON
FUEL SELECTOR	SWITCH TANK
ENGINE GUAGES	CHECK
MAGNETOS	L, R, THEN BOTH

NOTE: If operation is satisfactory on either magneto, continue on that magneto at reduced power and mixture full RICH to nearest airport. Prepare for power off landing.

CARBURETOR ICING

CARB HEAT	ON
MIXTURE	ADJUST FOR SMOOTHNESS

NOTE: Partial carb heat may be worse than no heat at all. It may melt some of the ice which will refreeze in the intake system. When using carb heat, always use full heat. When ice is removed, return the control to the full off position.

OPEN DOOR

TO CLOSE DOOR IN FLIGHT:

AIRSPEED	BELOW 89K1S
CABIN VENTS	CLOSE
STORM WINDOW	OPEN
IF UPPER LATCH IS OPEN	LATCH
IF SIDE LATCH IS OPEN	PULL ON ARMREST
	CLOSE LATCH
IF BOTH LATCHES ARE OPEN	LATCH SIDE
	THEN TOP

DELOVA CONTO

NOTE

IF BOTH UPPER AND LOWER LATCHES ARE OPEN, THE DOOR WILL TRAIL SLIGHLY OPEN AND AIRSPEEDS WILL BE REDUCED.

SPIN RECOVERY

	IDLE
AILERONS	NEUTRAL
RUDDER	FULL OPPOSITE
ELEVATOR	FORWARD TO BREAK STALL
	live and return to level flight.

CO DETECTOR WARNING

CO RST SOFT KEY	PRESS
IF WARNING CONTINUES	
CABIN HEAT	OFF
FRESH AIR SOURCE	OPEN
Land as soon as possible.	

ALTERNATOR FAILURE WITHOUT STANDBY ALT ANNUNCIATOR LIGHT ILLUMINATED:

CHECK TO VERIFY
OFF
NIMUM:
KER
CHECK/RESET
ON

NOTE

ALTERNATOR SWITCH......OFF

If alternator output cannot be restored, reduce electrical loads and land as soon as practical. The battery is the only remaining source of electrical power. Reference the aircraft POH for specific load shedding techniques.

Execute Complete Electrical Failure Checklist if battery is depleted.

FAILURE OF PRIMARY ALTERNATOR

INDICATIONS: ALT INOP ANNUNCIATOR. STBY ALT ON ANNUNCIATOR, ZERO AMMETER OUTPUT

STBY ALT	VERIFY ON
ENGINE RPM	INCREASE TO 2500
ELETRICAL LOAD	REDUCE BELOW 20 AMPS

NOTE

- If STBY ALT ON is flashing, reduce electrical loads until annunciator no longer flashes.
- The standby alternator is limited to 20 amperes continuous output. Transient operations of greater than 20 amperes for no more than 5 consecutive minutes may be conducted

ALTERNATOR OF
ALT FIELD CIRCUIT BREAKER CHECK AND RESE
ALTERNATOROI
If primary alternator power is not restored:
ALTERNATOROF
Land as soon as practical.

STANDBY ALTERNATOR

FAILURE OF STANDBY ALTERNATOR

STBY ALT FIELD CIRCUIT BREAKER
CHECK AND RESET
STBY ALT SENSE CIRCUIT BREAKER
CHECK AND RESET
STANDBY ALTERNATOR ON
If standby alternator power is not restored:
STANDBY ALTERNATOR OFF
If the standby alternator has failed or cannot provide
adequate power, the electrical power is dependent on
available battery storage.

Execute Complete Electrical Failure Checklist when battery depleted.

Reduce electrical loads by switching OFF or pulling circuit breakers for all non-essential equipment to include the

following:

- Reduce PFD/MFD Brightness
- Pitot heat
- Landing light
- Cabin lights
- Strobe lights
- Nav lights
- No. 2 nav/com/GPS

Land as soon as practical.

ELECTRICAL OVERLOAD (ALTERNATOR OVER 20 AMPS) Reduce electrical loads by switching OFF or pulling **ABOVE KNOWN ELECTRICAL LOAD.)**

ALTERNATOR SWITCH......ON BATTERY MASTER SWITCH OFF

IF ALTERNATOR LOADS ARE REDUCED:

ELECTRICAL LOAD REDUCE TO MINIMUM

LAND AS SOON AS PRACTICAL

IF ALTERNATOR LOADS ARE NOT REDUCED:

BATTERY MASTER SWITCHON ALTERNATOR SWITCH...... OFF

LAND AS SOON AS POSSIBLE

IF EQUIPPED WITH STANDBY ALTERNATOR

STBY ALTR..... VERIFY ON/CHECK AMMETER

NOTE

If STBY ALTR ON annunciator is flashing then reduce electrical loads until the annunciator no longer flashes.

If the standby alternator has failed or cannot provide adequate power, then electrical power is dependent on available battery storage. Duration of battery power available will be dependent on electrical load and battery condition prior to failure. Execute **COMPLETE ELECTRICAL FAILURE** checklist when battery is depleted.

WARNING

Compass errors may exceed 10° with both alternators inoperative.

CAUTION

Any power interruption will result in loss of attitude information from the PFD until the unit can be reinstated on the ground.

NOTE

LOW BUS VOLTAGE annunciator will be illuminated. Anticipate complete electrical failure. Duration of battery power available will be dependent on electrical load and battery condition prior to failure.

circuit breakers for all non-essential equipment to include the following:

- Reduce PFD and MFD brightness as part of overall electrical system management
- Pitot heat (unless Required)
- Landing light
- Cabin lights
- Strobe lights
- Nav lights
- No. 2 nav/com/GPS

LAND AS SOON AS PRACTICAL AND ANTICIPATE COM-PLETE ELECTRICAL FAILURE

EMERGENCY BATTERY VOLTAGE INDICATION

NOTE

Complete electrical failure is imminent. Use sound ADM to determine the best location for landing.

COMPLETE ELECTRICAL FAILURE

STANDBY INSTRUMENT VERIFY ON

CAUTION

The STBY PWR annunciator will rapidly flash for approximately one minute when aircraft power is lost. STBY PWR must be selected, otherwise the standby attitude indicator will auto shutdown after approximately one minute. Maintain aircraft control with reference to the standby airspeed, altimeter, and attitude indicators.

EMER BATT	ARM
BATTERY MASTER SWITCH	OFF
ALTERNATOR	OFF

PRIOR TO DESCENT:

VIIX I UKE	RICH
THROTTLE	SET FOR 500 FPM DESCENT
AT	DESIRED SPEED OR 126KTS

LAND AS SOON AS PRACTICAL

PITOT HEAT FAILURE

SEE APPROVED AIRCRAFT POH FOR THE **FOLLOWING CHECKLISTS:**

ERRONEOUS/LOSS OF CAS MESSAGES DUAL GPS FAILURE AVIONICS COOLING FAN FAILURE ELECTRIC TRIM FAILURE

PIPER SEMINOLE CHECKLIST

NOTE

Prior to beginning the preflight walk around, check oil fuel and notify line service if needed.

COCKPIT INSPECTION

AIRCRAFT DOCUMENTS	
TOW BAR/BAGGAGE	SECURE
CONTROL YOKE	RELEASE
PARKING BRAKE	SET
COMPASS DEV CARD	
CIRCUIT BREAKERS	CHECK IN
PITOT/STATIC	
ALTERNATE STATIC	
LANDING GEAR SELECTOR	DOWN
MAGNETO SWITCHES	OFF
ELECTRICAL SWITCHES	OFF
AVIONICS	OFF
COWL FLAPS	OPEN
FUEL SELECTORS	ON
BATTERY/MASTER	ON
FUEL QUANITITY GUAGES	CHECK
LANDING GEAR LIGHTS	3 GREEN
ANNUN PANEL LIGHTS	
FLAPS	EXTEND/40°
EXTERNAL LIGHTS/PITOT HEAT	ON/CHECK
PITOT HEAT	OFF
EXTERNAL LIGHTS	OFF
BATTERY/MASTER	OFF
EMPTY SEATS	FASTEN BELTS
WINDOWS/WINDSHEILD	CHECK/CLEAN
EMERGENCY EXIT	CHECK SECURE

CAUTION

Heated pitot tube becomes very hot. Ground operation should be limited to 3 minutes to avoid damage the heating elements.

CAUTION

If the emergency exit is unlatched in flight, it may separate and damage the exterior of the airplane.

NOTE

Set the parking brake by first pressing and holding the toe brake pedals and the pulling out the PARK BRAKE knob. The static drains are located on the lower left sidewall adjacent to the pilot.

PREFLIGHT INSPECTION

RIGHT WING/RIGHT ENGINE	
FUEL SUMPS	DRAIN
FLAPS & HINGES	CHECK
AILERONS & HINGES	CHECK
STATIC WICKS	CHECK
WING SURFACES	CHECK
WING TIP AND LIGHTS	CHECK
LEADING EDGE	CHECK
OIL QUANTITY	CHECK(6-8QTS)

FUEL TANK CHECK QUANTITY
FUEL VENTS...... CLEAR

NOSE SECTION

GENERAL CONDITION	CHECK
AIR INLETS	CLEAR
HEATER AIR INLET/EXHAUST	CHECK
BATTERY VENTS	CHECK
GEAR WELL AND DOORS	CHECK
GEAR LINES AND SWITCHES	CHECK
NOSE WHEEL TIRE	CHECK
NOSE GEAR STRUT	APPROX 2.7"
CHOCK	REMOVE

LEFT WING/LEFT ENGINE

LEFT WING/LEFT ENGINE	
OIL QUANITITY	CHECK(6-8QTS)
ALTERNATOR BELT	CHECK TENSION
PROP AND SPINNER	CHECK
FUEL TANK	CHECK QUANTITY
SCUPPER DRAIN	CLEAR
BRAKE PAD & DISC	CHECK
TIRE	CHECK
MAIN GEAR STRUT	APPROX 2.6"
GEAR LINES & SWITCHES	CHECK
GEAR WELL & DOORS	CHECK
CRANKCASE BREATHER TUBE .	
TIEDOWN/CHOCK	REMOVE
FUEL VENTS	
LEADING EDGE	
STALL WARNING VANES	CHECK
PITOT/STATIC MAST	UNOBSTRUCTED

LEFT WING/LEFT ENGINE CONTINUED

WING TIP & LIGHTS	CHECK
WING SURFACES	CHECK
AILERON & HINGES	CHECK
STATIC WICKS	CHECK
FLAP & HINGES	CHECK

NOTES

Eight quarts of oil are required for maximum range flights.

FUSELAGE (LEFT SIDE)

GENERAL CONDITION	CHECK
EMERGENCY EXIT	CHECK
ANTENNAS	CHECK
FRESH AIR INLET	CHECK
UNDERBELLY	CHECK

EMPENNAGE

CLIECK

GENERAL CONDITION	CHECk
FRESH AIR INLET	CLEAF
RUDDER	CHECk
STABILATOR AND TRIM TRAB	CHECk
STATIC WICKS	CHECk
TAIL CONE & SKID	CHECK FOR DAMAGE
TIE DOWN	REMOVE

FUSELAGE (RIGHT SIDE)

GENERAL CONDITION	CHECk
BAGGAGE DOOR	CHECk
CABIN DOOR	CHECk

STARTING ENGINE

BEFORE STARTING ENGINE

PREFLIGHT CHECK	COMPLETED
FLIGHT PLANNING	COMPLETED
PASSENGER BRIEFING	COMPLETE
CABIN DOOR	CLOSED & LATCHED
	(UPPER & LOWER)
SEATS	ADJUSTED & LOCKED
SEAT BELTS	FASTEN
PARK BRAKE	SET
GEAR	DOWN
THROTTLES	IDLE
PROP	FULL INCREASE
MIXTURES	CUT-OFF
FRICTION HANDLE	AS DESIRED
CARB HEAT/ALT AIR	OFF
COWL FLAPS	OPEN
STAB & RUDDER TRIM	SET
FUEL SELECTORS	ON
LEFT/RIGHT ALTR SWITCHES	ON
EMER BATT	ARM
	*VERIFY E VOLTS >23.3
AVION MASTER SWITCH	OFF
DAY/NIGHT SWITCH	AS REQUIRED
STROBE LIGHTS	FIN STROBE
ALL OTHER ELECTRICAL SWIT	CHESOFF
CABIN HEAT SWITCH	OFF
CIRCUIT BREAKERS	

NOTES

The EMERG BATT should remain ON after check for proper bus operation, allowing the PFD to remain powered for engine start. Avoid delays between this check and engine starting to reserve emergency battery power.

* If the EBATT VOLTS indication is less than 23.3 VOLTS, the voltage should be checked again at the end of the GROUND CHECK checklist (after being charged for some time by the primary electrical system). If EBAT VOLTS is still less than 23.3 volts, refer to Section 2.27 prior to flight.

ENGINE START—GENERAL

WARNING

The L START ENGD or R START ENGD warning CAS message will illuminate after 30 seconds of continuous engine cranking. If the CAS message illuminates after engine is running, stop the engine and determine the cause.

When starting at ambient temperatures +20 °F and below, operate first engine started with alternator ON (at max charging rate not to exceed 1500 RPM) for 5 minutes minimum before initiating start on second engine.

NOTE

Starter manufacturer recommends starter cranking periods be limited to 10 seconds with a 20 second rest period between cranking attempts. Maximum of 6 starts periods allowed. If a start is not achieved on the 6th attempt, allow starter to cool for 30 minutes before attempting additional starts. Do not engage the starter immediately after releasing it. This practice may damage the starter mechanism.

NORMAL START

NOMINAL STAIL	
BATT MASTR SWITCH	ON
GEAR POSITION INDICATORS	3 GREEN
PROPELLERS	FULL INCREASE
PROPELLER AREA	CLEAR
FUEL PUMPS	ON
MAGS LEFT/RIGHT	ON
Carbureted	
THROTTLE	0.25" OPEN
MIXTURE	RICH
PRIMER	AS REQ'D
STARTER	ENGAGED
Fuel Injected — Cold	
THROTTLE	0.25" OPEN
MIXTURE	. PRIME THEN CUT OFF
STARTER	ENGAGE
MIXTURE	ADVANCE
Fuel Injected — Hot	
THROTTLE	0.5" OPEN
MIXTURE	CUT OFF
STARTER	ENGAGE
MIXTURE	ADVANCE
IF ENGINE DOES NOT ST	ΓART PRIME
THROTTLE	1000RPM
OIL PRESSURE	CHECK

NOTE

When the engine starts, adjust the throttle and monitor oil pressure. If no oil pressure is indicated within 30 seconds, shut down the engine and have it checked. In cold weather it may take longer for an oil pressure indication.

AFTER START

VOLTS	CHECK
ALTR AMPS	CHECK
ANNUNCIATOR PANEL LIGHTS	OUT
GYRO VACUUM	4.8 TO 5.2 IN
HEATER	AS REQUIRED
FUEL PUMPS	OFF

ENGINE START WHEN FLOODED

*1:
FULL OPEN
CUT OFF
OFF
ENGAGE

WARM-UP

HROTTI ES	 1000-	1200	DDI/
ULCITES	 1000—	1200	LLINI

BEFORE TAXIING	
EXTERNAL POWER SOURCE	VERIFY REMOVED
BATT MASTR SWITCH	
FUEL SELECTORS	X-FEED(30 SECONDS)
AVION MASTER SWITCH	ON
MFDCHECK D	DATABASE CURRENCY
FUEL TOTALIZER	
CAS/PFD MESSAGES	
MSG SOFTKEY	CHECK
MASTER WARN & CAUTION	TEST
STANDBY INSTRUMENTS	CHECK
LIGHTS	
CABIN HEAT	
AUTOPILOTVERIFY SELF T	EST/MANUALLY TEST
FUEL SELECTORS	
ATIS/AWOS	
GPS/NAV/AP	SET
TAXI BRIEF	
ALTIMETER	SET
STANDBY ALT	
TRANSPONDER	ALT
MFD SET	AIRPORT DIAGRAM
DARKING BRAKE	DELEVCE

TAXIING

TAXI AREA	CLEAR
THROTTLES	APPLY SLOWLY
BRAKES	CHECK
STEERING	CHECK
INSTRUMENTS	CHECK

NOTE

During taxi, if the VOLTS indication decreases into the warning range, increase engine RPM (if possible) to retain adequate battery charging.

GROUND CHECK

PARKING BRAKESET		
MIXTURESFULL RICH		
PROPELLERSFULL INCREASE		
ENGINE INSTRUMENTSCHECK		
THROTTLES1500 RPM		
PROPELLERSFEATHER CHECK (500 MAX DROP)		
THROTTLES2000 RPM		
LEFT/RIGHT MAGCHECK		
MAX DROP175 RPM		
MAX DIFFERENCE50 RPM		
PROPELLERS EXERCISE(3x)		
GOVERNOR CHECKFIRST FLIGHT OF DAY		
CARB HEAT/ALT AIRCHECK		
THROTTLESIDLE CHECK		
FUEL PUMPSON		
THOTTLES1000 RPM		
FRICTION HANDLESET		
TAKE OFF BRIEFCOMPLETE		
NOTE		

Avoid prolonged ground operation with the carburetor heat ON as the air is unfiltered.

IF E-VOLTS WERE LESS THAN 23.3:

E-VOLT CHECK

BATT MASTER	OFF
ALTERNATOR SWITCH	OFF
E-VOLTS	CHECK
BATT/ALTERNATOR SWITCHES	ON

BEFORE TAKEOFF

DEI ONE TAKEOTT	
FLIGHT CONTROLS	FREE AND CORRECT
FLIGHT/ENGINE INSTRUMENTS	CHECK
FUEL QTY	SUFFICIENT
PROPELLERS	FULL INCREASE
MIXTURES	FULL RICH
CARB HEAT/ALT AIR	OFF
COWL FLAPS	OPEN
FLAPS	CHECK & SET
STAB AND RUDDER TRIM	SET
FUEL SELECTORS	ON
CAS MESSAGES	CHECK
CONTINUED IN NEXT COLUMN.	

PFD ANNUCIATIONS	CHECk
MSG SOFTKEY	CHECK
TRANSPONDER	ALT/CODE SET
GPS/NAV/AP	CHECK/SET
FUEL PUMPS	
PITOT HEAT	AS REQUIRED
STROBE LIGHTS	
LANDING/RECOG LIGHTS	
DOOR	UPPER/LOWER LATCHED
TIME OFF	NOTE
EINIAI ADDDOACH ADEA	CLEAR

PARKING BRAKE RELEASE CAUTION

Prior to take off, verify that the autopilot servos are disengaged and that flight controls move freely.

CAUTION

Fast taxi turns immediately prior to take off should avoided to prevent unporting fuel feed lines.

Adjust mixture prior to takeoff at high elevations. Do not overheat engines. Adjust mixture enough to obtain smooth engine operation.

NORMAL TAKEOFF

BRAKES	HOLD
THROTTLE	2000 RPM
ENGINE GAUGES	IN LIMITS
THROTTLE	FULL
BRAKES	RELEASE
ROTATE	75 KIAS
CLIMB	88 KIAS
POSITIVE RATE	GEAR UP
FLAPS	UP

NOTE

TAS aural alerts will be muted when GPS altitude is lower than 400 FT AGL.

Takeoff should not be attempted with ice or frost on the wings.

SHORT FIELD PERFORMANCE TAKEOFF

FLAPS	UP
STAB/RUDDER TRIM	CHECK/SET
BRAKES	HOLD
PROPELLERS	2700 RPM
THROTTLE	FULL
ENGINE GAUGES	CHECK
MIXTURE	FULL RICH (OR SET FOR ALT)
BRAKES	RELEASE
ROTATE	70 KIAS
CLIMB	82 KIAS
GEAR (POSITIVE RATE)	UP
OBSTACLE CLEAR	88KTS
	STAB/RUDDER TRIM BRAKES PROPELLERS THROTTLE ENGINE GAUGES MIXTURE BRAKES ROTATE CLIMB GEAR (POSITIVE RATE)

REPEAT FOR SECOND ENGINE

HOBBS/TACH.....RECORD FLIGHT CONTROLS......SECURE CABIN INTERIORCLEAN MIXTURECUT OFF MASTER SWITCHES OFF MAGS OFF DOOR/WINDOWCLOSED/LATCHED TIE-DOWNS/CHOCKS......SECURE BLANKET.....ON POST—FLIGHT INSPECTION......COMPLETE

SECURING AIRCRAFT

CLIMB CHECK—ABOVE 1000' AC	<u>3L</u>
GEAR	
FLAPS	RETRACTED
CLIMB POWER	25"MP/2500 RPM
CRUISE CLIMB AIRSPEED	105 KIAS
GPS/AUTOPILOT	
AREA	CLEAR
CRUISE CHECK	
CRUISE POWER	SET
PROPELLERS	RPM SET
MIXTURES	LEAN AS REQUIRED
FUEL PUMPS	OFF
LANDING LIGHT	OFF
RECOG LIGHT	OFF
COWL FLAPS	
HEADING INDICATOR	
PRE-MANUEVER CHECK	
MIXTURES	SET
FUEL PUMPS	ON
LANDING LIGHT	ON
RECOG LIGHT	ON
MIN. SAFE ALT	VERIFY
AREA	CLEARING TURNS
RADIO CALL	COMPLETE
POST MANUEVER CHECK	
LANDING LIGHT	OFF
RECOG LIGHT	
MIXTURES	AS REQUIRED
ENGINE GUAGES	CHECK
IN—RANGE CHECK/DESCENT	
WEATHER	
INSTRUEMNTS	
RADIOS	
ENVIRONMENT	
ALTIMETER	
HEADING INDICATOR	
MIXTURE	
CARB HEAT/ALT AIR	
LANDING LIGHTS	
RECOG LIGHTS	
DESCENT POWER	SET

IOLE	
APPROACH AND LANDING	
SEAT BACKS	
SEAT BELTS/HARNESSES	
FUEL PUMPS	
FUEL SELECTORS	
GEAR	
GEAR DOWN INDICATORS .	
NOSE GEARV	
MIXTURES	FULL RICH
PROPELLERS	FULL FORWARD
CARB HEAT/ALT AIR	AS REQUIRED
COWL FLAPS	OPEN
AUTOPILOTDISCON	NECTED ABOVE 200' AGL
FINAL GEAR CHECK	
500' AGL	GEAR VERIFIED DOWN
NORMAL LANDING	
FLAPS	0° TO FULL
AIRSPEED (FLAPS UP)	80-90 KIAS
AIRSPEED (FLAPS DOWN)	
TRIM	AS REQUIRED
THROTTLES	AS REQUIRED
TOUCHDOWN	MAIN WHEELS
BRAKING	AS REQUIRED
SHORT FIELD PERFORMAN	CE LANDING
FLAPS	
AIRSPEED	
TRIM	
THROTTLES	
TOUCHDOWN	
BRAKING MAXIN	
GO—AROUND	
MIXTURES	
PROPELLERS	
THROTTLES	
PITCH	
FLAPSR	
GEAR (POSITIVE RATE)	
COWL FLAPS	OPEN

AFTER LANDING CHECK	
RADIO CALL REPORT CLEAR	
FLAPS RETRACT	
COWL FLAPS AS REQUIRED	
CARB HEAT/ALT AIROFF	
FUEL PUMPSOFF	
LANDING LIGHT AS REQUIRED	
RECOG LIGHT AS REQUIRED	
PITOT HEATOFF	
HEATER	
FAN 2 MINUTES (IF ON FOR LANDING)	
FLIGHT PLANCLOSE	
TRANSPONDERALT	
MFDAIRPORT DIAGRAM	
TAXI CLEARANCEOBTAIN	
ENGINE SHUTDOWN	
HEATER (IF ON)FAN 2 MIN THEN OFF	
VENT FANOFF	
ENGINE TACH TIME MFDRECORD	
AVION MASTEROFF	
EMERG BATTOFF	
LEFT/RIGHT ALTROFF	
LEFT/RIGHT FUEL PUMPVERIFY OFF	
ALL OTHER ELETRICAL EQUIPOFF	
MIXTURESCUT OFF	
THOTTLESIDLE	
PANEL LIGHTSOFF	
LEFT/RIGHT MAG SWITCHESOFF	
BATT MASTEROFF	
STANDBY INSTRUMENT	
VERIFY SHUTDOWN	
NOTE	

NOTE

The flaps must be fully retracted for them to be safe to step on for aircraft exit.

In case the Aspen EFD-1000 standby instrument remains "ON" due to improper shutdown, the EFD-1000 switches to internal battery and depletes it. To turn off the EFD-1000, press the "SHUT DOWN" command from the Main Menu page 6 or hold the red "REV" button for 20 seconds.

EMERGENCY PROCEDURES

AIRSPEEDS FOR SAFE OPERATION		
V _{MC}	56 KIAS	
V _{XSF}	82 KIAS	
V _{YSE}	88 KIAS	
V. (3800lhs)	115 — 135 KIAS	

IF INSUFFICIENT RUNWAY REMAINS

MIXTURES	CUTOFF
FUEL SELECTORS	
MAGNETO SWITCHES	OFF
BATT MASTER	OFF
MAINTAIN DIRECTIONAL CONTROL TO A	AVOID OBSTA-
CLES	

ENGINE FAILURE DURING TAKEOFF (ABOVE 75 KIAS) -GEAR DOWN

DIRECTIONAL CONTROL	
THROTTLES	CLOSE IMMEDIATELY
LAND STRAIGHT AHEAD	
BRAKES	AS REQUIRED

ENGINE FAULURE DURNING TAKEOFF (ABOVE 75 KIAS) - GEAR UP/IN TRANSIT

WARNING

In many combinations of aircraft configuration, ambient conditions, and speed, negative climb performance may result. Refer to Climb Performance Chart—One Engine Operating—Gear Up, Figure 5-19.

PITCH	OO KIAC
DIRECTIONAL CONTROL	MAINTAIN
MIXTURES	FULL FORWARD
PROPS	FULL FORWARD
THROTTLES	FULL FORWARD
LANDING GEAR	CHECK UP
FUEL PUMPS	ON
FLAPS	FULL UP
FUEL SELECTORS	ON
INOP ENGINE	IDENTIFY AND VERIFY
THROTTLE (INOP)	CLOSE
PROP(INOP)	FEATHER
MIXTURE(INOP)	CUTOFF
BANK2° TO 3° TOW	ARD OPERATIVE ENGINE
TRIMA[DJUST FOR ZERO SIDESLIP

ENGINE SECURING

CLOSE
FEATHER
CUT OFF
CLOSE
OFF
OFF
OFF
OFF
REDUCE
AS REQUIRED
AREST AIRPORT

ENGINE FAILURE DURING FLIGHT

ENGINE ITALEONE DOMINO I EIGI	<u></u>
INOP ENGINE	IDENTIFY
THROTTLE	AS REQUIRED
AIRSPEED	AT LEAST 88 KIAS
FUEL PUMP	ON
MIXTURE	FULL RICH
FUEL QUANTITY	CHECK
CARB HEAT/ALT AIR	ON
OIL PRESSURE	CHECK
OIL TEMPERATURE	CHECK
MAG SWITCHES	CHECK
AIR START	ATTEMPT
If engine does not start, comple	ete ENGINE SECURING
LAND AS SOON AS PRACTICAL A	AT NEAREST AIRPORT

ONE ENGINE INOP LANDING

	INOP ENGINE	SECURED
	SEATBELT/HARNESS	SECURED
	FUEL SELECTOR(OP ENGINE)	ON
	MIXTURE(OP ENGINE)	FULL RICH
	PROP(OP ENGINE)	FULL FORWARD
	FUEL PUMP(OP ENGINE)	ON
	COWL FLAP(OP ENGINE)	AS REQUIRED
	ALTITUDE/AIRSPEED	MAKE NORMAL
į		APPROACH

	APPROACH
WHEN LANDING ASSURED:	
LANDING GEAR	DOWN
FLAPS	25°
FINAL APPROACH SPEED	90 KIAS
POWER	. RETARD SLOWLY AND
	FLARE AIRPLANE
TRIM	AS POWER IS REDUCED
AIRPLANE V	VILL YAW IN DIRECTION
	OF OPERATIVE ENGINE

WARNING

Under many conditions of loading and density altitude a go-around may be impossible and in any event a sudden application of power during one engine inoperative operation makes control of the airplane more

ONE EINGINE INOP GO-AROUND

MIXTURE	FOWARD
PROP	FOWARD
THROTTLE	T/O POWER
FLAPS	RETRACT SLOWLY
LANDING GEAR	RETRACT (POSITIVE RATE)
AIRSPEED	88 KIAS
TRIM	ADJUST 2°-3°
COWL FLAP(OP)	AS REQUIRED

EMERGENCY DESCENT

THROTTLES	CLOSED
PROPELLERS	FULL INCREASE
MIXTURES	AS REQUIRED
GEAR	DOWN BELOW 140KTS
AIRSPEED	135KTS

UNFEATHERING PROCEDURE(ACCUMULATOR)

FUEL SELECTOR(INOP)	ON
MAG SWITCHES(INOP)	ON
FUEL PUMP(INOP)	ON
MIXTURE	FULL RICH
THROTTLE	OPEN 1/4 INCH
PROP	FULL FORWARD
THROTTLE	REDUCE POWER
	UNTIL ENGINE IS WARM
ALTERNATOR	ON

NOTE

Starter assist is required if the propeller is not windmilling freely with 5-7 seconds after to prop control has been moved forward.

When prop unfeathering occurs, it may be necessary to retard the prop control slightly to not overspeed the prop.

LINEFATHERING PROCEDURE(STARTER ASSIST)

ONFLATHENING PROCEDO	JALISTANTEN ASSIST
FUEL SELECTOR(INOP)	ON
MAG SWITCHES(INOP)	ON
FUEL PUMP(INOP)	ON
MIXTURE	FULL RICH
Carbureted	
THROTTLE	TWO FULL STROKES
	THEN OPEN 1/4 INCH
Fuel Injected	
THROTTLE	OPEN 1/4 INCH
PROP	FORWARD
STARTERENGAG	E UNTIL PROP WINDMILLS
THROTTLEREDUCE	UNTIL ENGINE IS WARM
If engine does not start, p	rime as required.

ALTERNATORON

ENGINE ROUGHNESS

CADD !!EAT/A!T A!D

NOTE

Partial carburetor heat may be worse than no heat at all, since it may melt part of the ice which will refreeze in the intake system. Therefore, when using carburetor heat always use full and, when ice is removed, return the control to the full cold position.

CARB HEAT/ALT	AIRON	
If roughness continues after one minute:		
CARB HEAT	OFF	
MIXTURE	ADJUST FOR SMOOTHNESS	
FUEL PUMP	ON	
ENGINE GUAGE	S CHECK	
MAG SWITCHES	CHECK	

If operation is satisfactory on either magneto, continue on that magneto at reduced power and full rich mixture to the first airport.

ENGINE OVERHEAT

COWL FLAPS	OPEN
MIXTURE	ENRICHEN
POWER	REDUCE
AIRSPEEDINCREASE	ALTITUDE PERMITTING

HIGH OR LOW OIL PRESSURE

OIL PSI	VERIFY LOSS &
	AFFECTED ENGINE
THROTTLE M	INIMUM REQUIRED
PROPELLER	DECREASE
IF REQUIRED	
ENGINE SE	CURING CHECKLIST

CO DETECTOR WARNING

CO RST SOFT KEY	PRES
IF WARNING CONTINUES	;
CABIN HEAT	OF
FRESH AIR SOURCE	OPEI
Land as soon as possible	

FNGINE	FIRE DURIN	NG START

If engine has not started:

MIXTURE	CUTOFF
THROTTLE	FULL OPEN
STARTER	CONTINUE CRANKING

If engine has already started and is running, continue operating to try pulling the fire into the engine.

If fire continues:

FUEL SELECTORS	
FUEL PUMPS	OFF
MIXTURES	CUTOFF
THROTTLES	FULL OPEN
EXTERNAL FIRE EXTINGUISHER	USE
AIRPLANE	EVACUATE

NOTE

If fire continues, shut down both engines and evacuate

ENGINE FIRE IN FLIGHT

ELIEL CELECTOD/ACCECTED ENGINE)

FUEL SELECTOR(AFFECTED ENGINE)OFF
THROTTLE(AFF ENG)IDLE
PROP(AFF ENG)FEATHER
MIXTURE(AFF ENG)CUT-OFF
COWL FLAP(AFF ENG)OPEN
If fire persists:
EMERGENCY DESCENTINITIATE
If fire goes out:
AFFECTED ENGINESECURING CHECKLIST

Land as soon as possible at nearest airport.

FLECTRICAL FIRE

EEECTRICATETIALE	
FLASHLIGHT(NIGHT)	LOCATE
EMERG BATT	ARM
BATT MASTER	OFF
ALT SWITCHES	OFF
ALL ELECT SWITCHES	OFF
AVION MASTER SWITCH	OFF
VENTS	CLOSED
CABIN HEAT	OFF

If fire persists, locate and, if practical, extinguish with portable fire extinguisher located on the console just aft of the two front seats.

CONTINUED ON NEXT COLUMN

ELETRICAL FIRE (CONTINUED)

BUS TIE CIRCUIT BREAKERS

•	BOTH MAIN BUS	PULL
•	NON-ESSENTIAL BUS	PULL
•	AVIAONICS BUS #1	PULL
•	AVIONICS BUS #2	PULL
•	L. ALTERNATOR	PULL
•	R. ALTERNATOR	PULL
ALL	MAIN BUS CIRCUIT BREAKERS	PULL
ALL	AVIONICS BUS CIRCUIT BREAKERS.	PULL

NOTE

At this point, the pilot must decide if the flight can be safely continued without electrical power. If so, land at the nearest airport and have the electrical system repaired.

If electrical power is required for the safe continuation of the flight:

WARNING

The following procedure may reenergize the faulty system. Reset the circuit breakers one at a time. Allow a short time period between the resetting of each circuit breaker. If the faulty system is reinstated, its corresponding circuit breaker must be immediately pulled.

ONE MAIN BUS TIE CIRCUIT BREAKERIN
BATT MASTERON
L. OR R. ALT CIRCUIT BREAKERIN
L. OR R. ALT FIELD CIRCUIT BREAKERIN
ALT SWITCHON
MAIN BUS CIRCUIT BREAKERS
ELECTRIC TACHOMETERIN
GEAR INDICATORIN
AVIONICS BUS #1IN
AVIAONICS BUS #2IN
AVION MASTER SWITCHON
COMPASSIN
• AUDIOIN
• COMM #1IN
• NAV #1IN
VENTS OPEN(IF FIRE EXTINGUISHED)
Land as soon as practical.

WARNING

LANDING GEAR MUST BE LOWERED USING EMERGEN-CY GEAR EXTENSION PROCEDURE.

ONE ENGINE INOP FUEL MANAGEMENT

Using tuel tank same side as operating engine:	
FUEL SELECTOR(OPERATIVE)	ON
FUEL SELECTOR(INOPERATIVE)	.OFF
ELECTRUC FUEL PUMPS	.OFF

Using fuel from tank opposite operating engine:

FUEL SELECTOR(OPERATIVE)	CROSSFEED
FUEL SELECTOR(INOPERATIVE)	OFF
ELECTRIC FUEL PUMPS	OFF

NOTE

Use cross feed in level cruise flight only.

LANDING

FUEL SELECTOR(OPERATIVE)	N
FUEL SELECTOR(INOPERATIVE)O	FF

ENGINE DRIVE FUEL PUMP FAILURE

ELECTRICAL FUEL PUMP(AFF ENG)ON

FUEL QUANTITY

IF ONE TANK HAS LOW FUEL QUANTITY:	:
FUEL SELECTOR	XFEED
LAND AS SOON AS PRACTICAL	

IF BOTH TANKS HAVE LOW FUEL QUANTITY

FUEL SELECTORSON
LAND AS SOON AS POSSIBLE

LANDING GEAR UNSAFE WARNINGS

Red light indicates gear in transit.

Recycle gear if indication continues.

Light will illuminate and gear horn sounds when the gear is not down and locked if throttles are at low settings or wing flaps are in second or third notch position.

MANUAL GEAR EXTENSION

ing gear manually:
OFF
DAY
CHECK IN
ON
CHECK

TO EXTEND:

AIRSPEED	REDUCE (100 KIAS MAX
GEAR SELECTOR	DOWN
EMERG. GEAR EXTEND KN	NOB PUL
INDICATOR LIGHTS	3 GREEN
LEAVE EMERGENCY GEAR	EXTENSION KNOB OUT

SINGLE ALTERNATOR FAILURE

NOTE

Anytime total ties bus voltage is below 12.5Vdc, the LO BUS voltage annunciator will illuminate.

VERIFY FAILURE CHECK AMMETERS ELECTRICAL LOAD (IF LO BUS ILLUMINATED)

Reduce electrical load until load is less than 60 amps & voltage annunciator is extinguished

FAILED ALT SWITCHOFF
FAILED ALT CIRCTUIT BREAKER
CHECK AND RESET
FAILED ALT SWITCH(AFTER OFF AT LEAST 1 SECOND)
ON

If power not restored:

FAILED ALT SWITCH	OFF
AMMETER	MONITOR
N.	IAINITAINI DELOVAZEO ANADO

DUAL ALT FAILURE

VERIFY	AMMETERS
ELECTRIC LOAD	REDUCE TO MINIMUM
ALT SWITCHES	OFF
ALT CIRCUIT BREAKERS	CHECK/RESET
	AS REQUIRED
ALT SWITCHES	ON
(One at a time and off for a	nt least one second)

If only one alternator resets:

OPERATING ALT SWITCH.	ON
FAILED ALT SWITCH	OFF
ELECTRIC LOAD	REDUCE BELOW 60 AMPS
AMMETER	MONITOR

If neither resets:

BOTH ALT SWITCHES	OF
EMER BATT	VERIFY ARM
Continue flight on reduced electrical lo	oad on batter
power only.	

Land as soon as practical. Anticipate complete electrical failure. Duration of Battery power available will be dependent on electrical load and battery condition prior to failure. Consult the aircraft POH for load shedding procedure.

WARNING

If the battery is depleted, the landing gear must be lowered using the emergency gear extension procedure. The gear position lights will be inoperative.

COMPLETE ELECTRICAL FAILURE CAUTION

The emergency battery is designed to provide electrical power to all items on the emergency bus for a minimum of 30 minutes.

NOTE

The VOLTS indication on the EIS window automatically • Traffic changes to the emergency bus voltage (E VOLTS) when operating on the emergency battery.

NOTE

Cooling air for the PFD is lost when operating on the emergency bus as indicated by the PFD FAN FAIL CAS Advisory message.

EMERG BATT SWITCH	VERIFY ARM
STANDBY FLIGHT INST	VERIFY OPERATIONAL
AIRCRAFT CONTROL	USE PFD AND STANDBY
BATT MASTR SWITCH	OFF
ALTR LEFT/RIGHT SWITCHE	SOFF

PRIOR TO LANDING:

LANDING LIGHT	INOPERATIVE
GEAR	MANUAL GEAR EXTENSION
APPROXIMATELY 3	0 MINUTES OF ELECTRICAL POWER
IS AVAILABLE, LAND	D AS SOON AS POSSIBLE.

EMERGENCY BATTERY VOLTAGE

WARNING

COMPLETE ELECTRICAL FAILURE IS IMMINENT LAND AS SOON AS POSSIBLE

PFD FAILURE

· · · · · · · · · · · · · · · · · · ·	
STANDBY INSTRUMENT VERIFY OPERATIONAL	
AIRCRAFT CONTROLUSE STANDBY	
DISPLAY BACKUP (RED BUTTON) PUSH	
AIRCRAFT CONTROL USE MFD AND STANDBY	
COM 2ACTIVATE AND TUNE	
NAV 2ACTIVATE AND TUNE	
COM2/MICSELECT	
DMESELECT NAV2 IN DME TUNING WINDOW	
EXIT AND AVOID IMC AS SOON AS PRACTICAL	

NOTE

If the PFD fails, the MFD will remain in normal display mode. Pushing the DISPLAY BACKUP button on the audio panel puts the MFD in reversionary mode, which depicts primary flight instruments, engine and systems information on a single display format. Certain map functions will be lost in reversionary mode.

NOTE

If the PFD failure occurs while operating on NAV1 DME, the NAV 1 DME information will still be available. EXIT AND AVOID FLIGHT IN IMC If however, the pilot selects NAV2 DME, NAV1 DME may not be re-selected.

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PFD FAILURE CONT'D

NOTE

The following features will become inoperative if the PFD fails:

- Com 1 (red x'd but 121.5 MHz remains available)

- GPS 1
- GFC700 Autopilot

MFD FAILURE

DISPLAY BACKUP on audio panelPUSH **EXIT AND AVOID IMC AS SOON AS PRACTICAL**

NOTE

The PFD will automatically revert to reversionary mode.

NOTE

The following features will become inoperative with MFD failure:

- Com 2 (red x'd but 121.5 MHz remains available)
- GPS 2
- GDL 69 (Garmin Datalink XM) DME • ESP
- ADF

NOTE

If the GFC700 autopilot was engaged prior to MFD failure, it will remain engaged in its current lateral and vertical modes. The modes cannot be changed and if the autopilot is disengaged, it cannot be re-engaged.

ATTITUDE HEADING REFERENCE SYSTEM (AHRS) **FAILURES**

AHRS TOTAL FAILURE ON GROUND:

PFD ADVISORIES	CONSIDER
AHRS Circuit Breaker	RESET
IE AHRS DATA STILL INIVALID:	

IFR OPERATIONS NOT AUTHORIZED

IN FLIGHT:

STANDBY INSTRUMENT	
VERIFY NO FAILURE IN	DICATIONS
ATTITUE AND HEADINGUS	E STANDBY

NOTE

The course pointer will point upwards at all times, but the CDI can still be used for navigating in GPS and VOR/LOC modes.

COURSE SE	T USING CRS KNOB ON PFD
PFD MESSAGE ADVISORIE	SCONSIDER
AHRS Circuit Breaker	RESET

IF AHRS DATA STILL INVALID:

NOTES FOR THIS CHECKLIST FOUND AT TOP OF NEXT PAGE...

NOTE

The autopilot (including ESP and USP) will be inoperative if AHRS remains inoperative.

NOTE

Traffic (TAS) symbols will not be displayed on the moving map, however TAS information remains available on the TAS page.

NOTE

For partial AHRS failures, a red-x and amber text will appear over the affected indication(s)

AIR DATA COMPUTER (ADC)FAILURES

ADC TOTAL FAILURE

ON GROUND:

PFD MESSAGE ADVISORIES CONSIDER ADC Circuit Breaker.....RESET IF ADC DATA STILL INVALID:

IFR OPERATIONS NOT AUTHORIZED

IN FLIGHT:

STANDBY INST	VERIFY AIR	DATA INDICATIONS
AIRSPEED, ALTITUD	E, AND V/S	USE STANDBY
PFD MESSAGE ADV	ISORIES	CONSIDER
ADC Circuit Breaker		RESET

NOTE

During failure of ADC, Traffic Avoidance System (TAS) and GFC700 autopilot will be inoperative. IF ADC DATA STILL INVALID

EXIT AND AVOID FLIGHT IN IMC

ERRONEOUS INDICATION OR LOSS OF ENGINE AND FUEL DISPLAYS

NOTE

Erroneous information should be suspected when indications do not agree with other system information. Erroneous indications may be identified by comparing a display with other system information.

- 1. Set power based on throttle lever position, engine sound and speed.
- Monitor other indications to determine the health of the engine.
- Use known power settings from POH power setting tables for approximate fuel flow values.
- Use other system information, such as annunciator messages, fuel totalizer quantity and flow, to safely complete the flight.

IF INDICATIONS FOR ANY OF THE FOLLOWING ARE INVALID:

- All Engine Parameters
- VOLTS
- ALTR AMPS
- BATT AMPS
- FUEL QTY

GEA Circuit Breaker.....RESET IF ALL ENGINE PARAMETERS ARE STILL UNAVAILABLE,

LAND AS SOON AS PRACTICAL.

ERRONEOUS OR LOSS OF WARNING/CAUTION CAS **MESSAGES**

- 1. If a yellow-x is placed over the CAS message window, special attention should be placed on all engine and airframe related indications. The Master Warning and Master Caution indicators will not function, therefore a failure of a particular system can go undetected.
- If a CAS message appears that is not expected, treat it as if the condition exists.
- If an abnormal condition exists but the CAS system has not been activated, use other available information to confirm the condition exists. If it cannot be determined that the condition does not exist, treat the situation as if the condition does exist and take appropriate action.

NOTE

CAS messages are inhibited for many parameters on the engine information system display (EIS) of the MFD. The Master Warning and Master Caution indicators and associated chimes are still activated whenever any indicated parameter enters the red or amber color bands.

IF A RED X APPEARS OVER THE CAS MESSAGE WIN-DOW, LAND AS SOON AS PRACTICAL.

COMMUNICATIONS (COM1 AND COM2) FAILURE NOTE

No matter what the cause of a Com failure, removing power from the audio panel actuates a fail-safe connection between the pilot's headset/microphone and COM1.

AUDIO MKR Circuit BreakerPULL **FXIT AND AVOID IMC AS SOON AS PRACTICAL**

DUAL GPS FAILURE

NAVIGATION.....USE ALT. SOURCE

IF NO ALTERNATE NAV SOURCES ARE AVAILABLE

Dead Reckoning (DR) mode—Is active when in Enroute mode (the airplane is greater than 30NM from the destination airport in flight plan).

NAVIGATION...... USE MFD AND AMBER CDI ON HIS WARNING

In DR mode, the estimated position becomes increasingly unreliable over time and should not be used as a sole means of navigation. In DR mode the CDI is initially displayed in amber, but is removed after 20 minutes.

TAWS is Inoperative

DR mode uses heading, airspeed and last known GPS position to estimate the airplanes current position. All maps with an airplane symbol show a ghosted airplane and a "DR" label.

NOTE

Traffic Information System (TIS) and Traffic Advisory System (TAS) are not dependent on GPS information. Therefore, the position of displayed traffic relative to the airplane symbol on the map is still accurate.

Loss of integrity (LOI) Mode — Is active when GPS integrity is insufficient for the current phase of flight. NAVIGATION.....USE OTHER SOURCES

All information derived from GPS or DR is removed from the displays.

The airplane symbol is removed from all maps. The map will remain centered at the last known position. "NO GPS POSITION" is shown in the center of the map. TAWS and TAS are inoperative.

AUTOPILOT MALFUNCTION

CONTROL WHEEL	GRASP FIRMLY
ATTITUDE INDICATORS	CROSSCHECK
A/P DISC SWITCH	DEPRESS AND HOLD
PITCH TRIM	RETRIM
AUTOPILOT Circuit Breaker .	PULL
AUTOPILOT	DO NOT RE-ENGAGE

AUTOMATIC AUTOPILOT DISCONNECT

A/P DISC SWITCH	DEPRESS AND RELEASE
PITCH TRIM	RETRIM AS NECESSARY

NOTE

The autopilot disconnect may be accompanied by a red boxed PTCH (pitch), ROLL, or PTRM annunciation on the PFD, indicating the axis which has failed. The autopilot cannot be re-engaged with any of these annunciations present.

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ELECTRIC PITCH TRIM FAILURE

Loss of the electric pitch trim servo will not cause the autopilot to disconnect. Monitor pitch attitude for unusual behavior. Be alert to possible autopilot out-oftrim conditions (see AUTOPILOT OUT OF TRIM procedure) and expect residual control forces upon disconnect. The autopilot will not re-engage after disconnect with failed pitch trim.

AUTOPILOT...... DISCONNECT **ELECTRIC PITCH TRIM RUNAWAY**

After the autopilot is disengaged, it can not be reengaged until the electric pitch trim system regains functionality.

CONTROL WHEEL	GRASP FIRMLY
ATTITUDE INDICATORS	CROSSCHECK
A/P DISC SWITCH	DEPRESS AND HOLD
PITCH TRIM Circuit Breaker	PULL
PITCH TRIM	RETRIM MANUALLY

AUTOPILOT OVERSPEED RECOVERY

NOTE

This autopilot mode is activated whenever the aircraft actual or projected airspeed exceeds V_{NF}. THROTTLE..... REDUCE POWER AUTOPILOT PITCH REFERENCERESET AUTOPILOT...... DISCONNECT NOTE

Overspeed recovery mode provides a pitch up command (to a maximum level flight altitude) to decelerate the airplane below VNE. The autopilot must be en-

flight director Overspeed recovery is not active in altitude hold (ALT), glideslope (GS) of glidepath (GP) modes. The speed reference cannot be adjusted while in overspeed recovery mode.

gaged for it to follow the pitch-up commands of the

AUTOPILOT UNDERSPEED RECOVERY

NOTE

This autopilot mode is active whenever the autopilot is engaged and the airspeed has decreased below a minimum threshold.

THROTTLE	INCREASE POWER
FLAPS POSITION	CONSIDER
LANDING GEAR POSITION	CONSIDER

AUTOPILOT OUT-OF-TRIM

CAUTION

Do not attempt to overpower the autopilot in the event of a mistrim. The autopilot servos will oppose pilot input and will trim opposite the direction of pilot input (pitch axis only). This could lead to a significant out-of -trim condition. Disconnect the autopilot using the A/P DISC / TRIM INTER switch if manual control is desired.

IF AIL \rightarrow OR \leftarrow AIL annunciationVERIFY SLIP
SKID INDICATOR CENTERED
IF ↑ ELE OR ↓ ELE ANNUNCIATION
SUSPECT ELECATOR TRIM ISSUE
CONTROL WHEELGRASP FIRMLY

CAUTION

Be prepared to apply a sustained control force in the direction of the annunciation arrow. For example, an arrow pointing to the right with AIL annunciation indicates that sustained right wing down control wheel force will be required upon autopilot disconnect.

AP DISC SWITCH	DEPRESS
AFFECTED TRIM SYSTEM	RETRIM
AUTOPILOT	RE-ENGAGE IF AVAILABLE
If mistrim indication re-o	ccurs, disconnect autopilot for
the remainder of the fligi	ht or until the offending condi-
tion is resolved.	

ABNORMAL FLIGHT DIRECTOR MODE TRANSITIONS NOTE

Upon loss of a selected mode, the system will revert to the default mode for the affected axis, either ROL or PIT

CELECT DIFFERENT MODE

LOSS OF VERTICAL MODE:

ALITODILOT

	AUTOFILOTSELECT DITTERENT MODE	
i	IF ON AN INSTRUMENT APPROACH	
i	AUTOPILOT DISCONNECT	

LOSS OF LATERAL MODE:

AUTOPILOT	SELECT DIFFERENT MODE
IF ON AN INSTRUME	NT APPROACH
AUTOPILOT	DISCONNECT

AUTOPILOT PREFLIGHT TEST FAILURE

÷	AUTOPILOT Circuit Breaker	PULL
÷	PITCH TRIM Circuit Breaker	PULL
:	AUTO PILOT AND PITCH TRIM	RESET

LOSS OF NAVIGATION INFORMATION

AUTOPILOTSELECT	ANOTHER LATERAL MODE
NAV SOURCE	SELECT A VALID SOURCE
AUTOPILOT	SELECT NAV
IF ON AN INST. APP.	
MISSED APPROACH	EXECUTE

COOLING FAN FAILURES

IF FAILURE OCCURS ON THE GROUND DO NOT FLY UNTIL ISSUE IS RESOLVED IF FAILURE OCCURS IN FLIGHT

FIX ISSUE PRIOR TO NEXT FLIGHT

PITOT HEAT

PITOT HEAT	O
IF "PITOT HEAT FAIL" SHOWS:	
PITOT HEAT	OF
PITOT HEAT Circuit Breaker	RESE
PITOT HEAT	O
IF PITOT HEAT IS STILL INOPERATIV	/E:
IMC/ICING CONDITIONS	EXIT/AVOID

HADDVILLE BLIMB EVILLIBES

HTDRAULIC POWIP FAILURES
HYDRAULIC PUMP WILL NOT DEACTIVATE:
GEAR PUMP Circuit BreakerPULL
PRIOR TO LANDING
GEAR PUMP Circuit BreakerPUSH
IF CIRCUIT BREAKER POPS
MANUAL GEAR EXTENSIONCOMPLETE

HYDRAULIC PUMP WIL	L NOT ACTIVATE:
GEARRET	URN TO ORIGINAL POSITION
GEAR PUMP Circuit Bre	akerRESET
GEAR	RESELECT
IF LANDING GEAR REM	AINS UP
MANUAL GEAR EXTENS	ION COMPLETE
IF LANDING GEAR REM	AINS DOWN
AIRSPEED	BELOW 140 KIAS

STARTER ENGAGED

IF ON THE GROUND:

TH	ROTTLE	REDUCE
EN	G START Circuit Breaker	PULL
EN	GINE	SHUTDOWN

IF IN FLIGHT:

THROTTLE	REDUCE
ENG START Circuit Breaker	PULI
LAND AS SOON AS POSSIBLE	

SPIN RECOVERY (INTENTIONAL SPINS PROHIBITED) NOTE

FAA Regulations do not require spin demonstration of SIGNIFICANT CLIMB PERFORMANCE PENALTIES CAN multi-engine airplanes; spin tests have not been con- RESULT FROM LANDING GEAR, FLAP, OR WINDMILLducted. The recovery technique presented is based on I ING PROPELLER DRAG. THESE PENALTIES ARE APPROXthe best available information.

THROTTLES IDLE RUDDER......FULL OPPOSITE SPIN CONTROL WHEEL......FULL FORWARD AILERONS..... NEUTRAL RUDDER......NEUTRAL WHEN SPIN STOPS CONTROL WHEEL.....SMOOTH AFT TO RECOVER FROM DIVE

SUMMARY OF FACTORS AFFECTING SINGLE ENGINE **OPERATIONS**

IMATED AS FOLLOWS:

LANDING GEAR EXT./FLAPS UP-250FPM FLAPS EXTENDED 25°/GEAR DOWN.....-490FPM FLAPS EXTENDED FULLY/GEAR DOWN......-525FPM INOP ENG. WINDMILLING (GEAR AND FLAPS UP).....

To close door in flight:

AIRSPEED	SLOW TO 82 KIAS
CABIN VENTS	CLOSE
STRM WINDOW	OPEN
TOP LATCH(IF OPEN)	LATCH
SIDE LATCH(IF OPEN)	PULL AND LATCH

If both latches are open:

SIDE LATCH	LATCH
	THEN LATCH TOP LATCH

PROP OVERSPEED

THROTTLE(AFF ENG)	RETARD
OIL PRESSURE(AFF ENG)	CHECK
PROP CONTROL(AFF ENG)	DECREASE
THEN SET IF ANY C	ONTROL AVAILABLE
AIRSPEED	REDUCE
THROTTLE(AFF ENG)	AS REQUIRED

EMERGENCY EXIT

PLASTIC COVER	REMOVE
EMERGENCY EXIT HANDLE	PULL
WINDOW	PUSH OUT

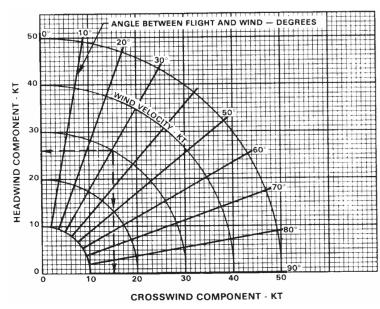
HEATER OVERHEAT

`ARINI HEAT	OFF

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NORTH STAR AVIATION **APPENDICIES**NORTH STAR AVIATION **APPENDICIES**

Appendix A Wind Component/Light Gun Signals



ATC LIGHT GUN SIGNALS FOR AIRCRAFT

COLOR & TYPE	GROUND	AIR
STEADY GREEN	Cleared for takeoff	Cleared to land
FLASHING GREEN	Cleared for taxi	Return for landing (to be followed by steady green)
STEADY RED	STOP!	Give way to other aircraft and continue circling
FLASHING RED	Taxi clear of runway in use	Airport unsafe, do not land
FLASHING WHITE	Return to starting point on airport	N/A
ALTERNATING RED/GREEN	Exercise extr	reme caution

Appendix B Maneuver Sheets



Piper Seminole Flight Maneuvers



North Star Aviation

Minnesota State University, Mankato

Slow Flight (Landing Configu	ration)	Slow Flight (Takeoff Confi	guration)
AltitudeRecovered by 3	000' AGL Minimum	AltitudeRecovered b	y 3000' AGL Minimun
Throttles	Below 15" MP	Power	Below 15" MF
Pitch to Maintain Altitude		Pitch to Maintain Altitude	
Below 140 KIAS	Gear Down	Below 100 KIAS	Props Forward
Below 111 KIAS	Flaps 10°	PowerIncre	ase to Maintain Altitud
Below 100 KIAS	Props Forward	PitchMai	ntain Stall Speed +10/-
Below 90 KIAS	Flap 25°		(Target 62 KIAS
Below 90 KIAS	Flap 40°	Trim	Adjust as Necessar
PowerIncrease	to Maintain Altitude	Recovery:	
PitchMainta	in Stall Speed +10/-0	Smoothly reduce pit	ch
	(Target 60 KIAS)	Power	Increase to Max
Trim	.Adjust as Necessary	Positive Rate	Verify Gear U
Recovery:			Verify Flaps 0
Smoothly Reduce Pitch		Accelerate to Cruise	Flight110 knot
Power	Increase to Max.		
Flaps	25°	Power-On Stall (Takeoff C	onfiguration)
Pitch for minimal loss of	of altitude	AltitudeRecovered b	y 3000' AGL Minimu
Positive Rate	Gear Up	Power	
	Flaps 10°	Pitch to maintain altitude	
	Flaps 0°	Below 100 KIAS	Props Forwar
Accelerate	V _Y (88 KIAS)	82 KIAS	
Power-Off Stall (Landing Cor	ifiguration)	Smoothly Increase Pitch to I	nduce a Stall
Altitude Recovered by 3	000' AGL Minimum	Recovery: (Initiated at the first	t indication for
Power	Below 15" MP	commercial)	
Pitch to Maintain Altitude		Smoothly reduce pitch	
Below 140 KIAS	Gear Down	Power	Increase to Max
Below 111 KIAS	Flaps 40°	Pitch for minimal lo	ss of altitude
	(one notch at a time)	Positive Rate	Verify Gear U
Below 100 KIAS	Props Forward		Verify flaps (
Enter Normal Descent to Land	•	Accelerate to Cruise	Flight110 knot
	Throttles to Idle		
Maintain Altitude to Induce a S	ta11	Steep Turns	
Recovery: (Initiated at the first in	ndication for	Altitude	3000' AGL Minimur
commercial)	•	Power18"	MP and 2300-2500RPM
Smoothly Reduce Pitch		Airspeed	110 KIA
Power	Increase to Max.	Bank	50
Flaps	25°	PowerIncrea	se to Maintain Airspee
Pitch for minimal loss of	of altitude	TrimRoll	Aft to Relieve Pressure
Positive Rate	Gear Up	Roll Out	Initial Headin
	Flaps 10°	Execute a 360° turn in the op	posite direction
<u></u>	Flaps 0°		
Accelerate to Cruise Fli	ght110 knots		

Revised: 8/27/2018

Piper Seminole Flight Maneuvers

Accelerated Stall	
AltitudeRecovered by 3000' AGL Minimum	
ThrottlesBelow 15" MP	
Pitch to Maintain Altitude	
Below 100 KIASProps Forward	
Bank45°	
Throttles	
Maintain altitude to induce stall	
Recovery: (Initiated at the first indication for	
commercial)	
Smoothly reduce pitch	
BankWings Level	
PowerIncrease to Max.	
Pitch for minimal loss of altitude	
Positive RateVerify Gear Up	
Verify flaps 0°	
Accelerate to Cruise Flight 110 knots	
<u>Drag Demonstration</u>	
Altitude4000' AGL Minimum	
ThrottlesBelow 15" MP	
Cowl FlapsAs Required	
Pitch to Maintain Altitude	
Below 100 KIASProps Forward	
Airspeed - 88 KIAS	
Operating EngineMax Power	
Inoperative EngineSim. Feather	
(11.5" MP, 2000 RPM)	
Airspeed 88kts. (VYSE)Note Performance	
Airspeed 78ktsNote Performance	
Airspeed 98ktsNote Performance	
Airspeed 88kts (Vyse)	
Gear DownNote Performance	
Flaps 10°Note Performance	
Flaps 25°Note Performance	
Flaps 40°Note Performance	
Inoperative Engine (windmilling)	
PropForward	
Throttle	
Note Performance	
Flaps 25°	
Flaps 10°	
Flaps 0°Note Performance	
Gear UpNote Performance	
Inoperative Engine15" MP (warm-up CHT)	
Accelerate to Cruise Flight110 knots	

V _{MC} Demonstration
AltitudeRecovered by 4000' AGL Minimum
ThrottlesBelow 15" MP
Below 100 KIASProps Forward
Airspeed – 92 KIAS
Left engineThrottle to Idle
Right engineThrottle to Full
PitchIncrease to Lose 1kt Per Second
Recover at the first indication of a stall
Or
Red Radial Line 56kts (V _{MC})
Or
Loss of Directional Control
Recovery:
Right EngineReduce to Regain Dir.
Control
Reduce Pitch (min. loss of alt.)
Directional Control Regained
Right EngineIncrease power to full
Pitch for Minimal loss of Altitude
AccelerateV _{YSE} (88 KIAS)
Inoperative Engine15" MP (warm-up CHT)
Accelerate to Cruise Flight
Ground Reference Maneuvers
Altitude600-1000'AGL
Pre-Maneuver CheckComplete
AreaClear of Obstructions
Airspeed
Enter Maneuver on Downwind Heading Perform to Applicable Test Standards
renorm to applicable rest standards
V-Speeds
V _Y = 88 KIAS Best Rate of Climb
V _{YSE} = 88 KIAS Singe Engine Best Rate of Climb
V _X = 82 KIAS Best Angle of Climb
V _{XSE} = 82 KIAS Single Engine Best Angle of Climb
V _{SSE} = 82 KIAS Min. Intentional One Engine Inop.
V _{SO} = 55 KIAS Stall Speed (Landing Configuration)
$V_S = 57 \text{ KIAS} \text{ Stall Speed (Clean Configuration)}$
V _{MC} = 56 KIAS Minimum Control
V _{FE} = 111 KIAS Maximum Flaps Extended
V _{LE} = 140 KIAS Maximum Landing Gear Extended
V _{LO Down} = 140 KIAS Max. Landing Gear Extension
VLOUP = 109 KIAS Max. Landing Gear Retraction
V _{NO} = 169 KIAS Maximum Structural Cruising
V _{NE} = 202 KIAS Never Exceed

Vo = 115 - 135 KIAS --- Maneuvering Speed (VA)

Piper Seminole Flight Maneuvers



Piper Warrior Flight Maneuvers

North Star Aviation Minnesota State University, Mankato



Altitude......2000' AGL Minimum

Power Off Stalls (Landing Configuration)

Slow Flight (Landing Configuration) Altitude......2000' AGL Minimum Pre-Maneuver Check......Complete Throttle......1700RPM Pitch to Maintain Altitude
Below 103 KIAS.....Flaps 40° (One Notch at a Time)
Throttle......Increase to Maintain Altitude (1800-2000 RPM) (1800-2000 RPM)
Pitch......Maintain Stall Speed +10/-0 (Target 49 KIAS) (Target 49 KIAS)
Trim.....Adjust As Necessary Recovery: Smoothly Reduce Pitch Throttle......Max Power Pitch for minimal loss of altitude Positive Rate.....Flaps 10° Accelerate.....Vy Return to Cruise Flight 90 KIAS Slow Flight (Takeoff Configuration) Altitude......2000' AGL Minimum Pre-Maneuver Check......Complete Pitch to Maintain Altitude Throttle......Increase to Maintain Altitude (1800-2000RPM) Pitch.....Stall Speed +10/0 (Target 55 KIAS) Trim.....Adjust as Necessary Recovery: Smoothly Reduce Pitch
Throttle......Max Power Pitch for minimal loss of altitude Positive Rate......Verify Flaps 0° Accelerate.....Vy Return to Cruise Flight 90 KIAS Ground Reference Altitude......600-1000'AGL Pre-Maneuver Check......Complete Area......Identify Pos. Landing Area Enter Maneuver on Downwind Heading Perform to Applicable Test Standards

Throttle to IdleMax Power25° udeFlaps 10°VY speed 90 KIAS ion) 00° AGL MinimumComplete
Throttle to IdleMax Power25° udeFlaps 10°VY speed 90 KIAS ion) 00° AGL MinimumComplete
ude
Flaps 10 ⁶ Vy speed 90 KIAS ion) 00' AGL MinimumComplete
Flaps 0° Vy speed 90 KIAS ion) 00' AGL Minimum
wyspeed 90 KIAS ion) 00' AGL Minimum Complete
ion) O' AGL MinimumComplete
ion) 00' AGL Minimum Complete
00' AGL Minimum Complete
00' AGL Minimum Complete
Complete
1500 RPM
Max Power
:a11
Max Power
TT :0 T4 00
Verify Flaps 0°
Vy
AS
00' AGL Minimum
Complete
2300-2400RPM
90 KIAS
Commercial (50°)
Private (45°)
lieve Back Pressure
Maintain Airspeed
itial Heading+/- 10°
tion as necessary
non as necessary

Revised: 12/26/2018 Piper Warrior Flight Maneuvers

Wallouvers

Revised: 8/27/2018

1

Eights on Pylons	Steep Spirals
Airspeed	Altitude
Pre-Maneuver Check	(Altitude enough to complete three turns. Consider DA)
AreaClear of Obstructions	Pre-Maneuver CheckComplete
Pivotal AltitudeGS ² /11.3	Airspeed90 KIAS
Enter 45° to Downwind	Begin Maneuver on Downwind Heading
PitchMaintain Pivotal Altitude	Prior to being abeam the reference point
Perform 2 revolutions, one around each point	Throttle
2 citim 2 reversions, one around each point	PitchVg
Lazy Eights	BankUp to 60°
Altitude2000' AGL Minimum	(Maintain Equal Radius)
Pre-Maneuver Check	Each Upwind Heading - Clear the engine by adding
Airspeed90 KIAS	power slowly up to 1700 RPM
(Increasing Pitch, Increasing Bank)	After 3rd Turn:
45° PointMax Pitch Up, 15° Bank	Wings Level
(Decreasing Pitch, Increasing Bank)	Heading +/- 10°
90° PointLevel Pitch, 30° Bank	Recovery
(Decreasing Pitch, Decreasing Bank)	Return to Cruise Flight
135° Point	Or
(Increasing Pitch, Decreasing Bank)	Climb As Assigned
(Increasing Fitch, Decreasing Bank) 180° Point	Or Or
Straight and Level	Proceed With Simulate Power-Off Landing
Initial Heading +/- 10°	
Initial Altitude +/- 100'	(No Lower than 500' AGL)
	*** ***
Initial Airspeed +/- 10 KIAS	Warrior V-Speeds
Repeat in Opposite Direction	V _R = 55 KIAS Rotate
Chandelles	Vy = 79 KIAS Best Rate of Climb
Altitude2000' AGL Minimum	Vx = 63 KIAS Best Angle of Climb
Pre-Maneuver Check	V _{SO} = 44 KIAS Stall Speed (Landing Configuration)
Airspeed 90 KIAS	V _S = 50 KIAS Stall Speed (Clean Configuration)
Bank 30°	
Throttle Max Power	VFE = 103 KIAS Maximum Flaps Extended Speed
Pitch Gradually Increase	V _{NO} = 126 KIAS Maximum Structural Cruising Speed
90° PointMax Pitch Up	V _{NE} = 160 KIAS Never Exceed Speed
Maintain Pitch, Gradually Decrease Bank	V _A = 88 - 111 KIAS Maneuvering Speed
180° Point	V _G = 73 KIAS Best Glide Speed
Roll Out+/- 10° Heading	•
Pitch	Archer V-Speeds
Slowly decrease pitch to accelerate while holding altitude	V _R = 60 KIAS Rotate
Return to Cruise Flight 90 KIAS	$V_Y = 76 \text{ KIAS} \text{ Best Rate of Climb}$
	$V_X = 64 \text{ KIAS} \text{ Best Angle of Climb}$
	V _{SO} = 45 KIAS Stall Speed (Landing Configuration)
	V _S = 50 KIAS Stall Speed (Clean Configuration)
	VFE = 102 KIAS Maximum Flaps Extended Speed
	V _{NO} = 125 KIAS Maximum Structural Cruising Speed
	V _{NE} = 154 KIAS Never Exceed Speed



Takeoff and Landing Procedures

North Star Aviation
Minnesota State University, Mankato



Piper Warrior/Archer

Normal Takeoff	_
Flaps	0°
Line up on runway centerline:	
Brakes	
Rwy/Compass/DGVerify al	
Brakes	
Throttle	
Engine Gauges	Verify in the green
"Airspeed Alive"	
"VR"	"Rotate
Pitch	
Trim	
Above 1000' AGL and Clear of Par	tternClimb Check
Soft-Field Takeoff	
Before Taking Runway	
F1aps	25
Flight Controls	
Full Aft Line up on runway cent	
Rwy/Compass/DGVerify al	ierinie without stopping
ThrottleVerify at	
Engine Gauges	
	verity in the green
"Airspeed Alive" Lift-Off	A
Reduce Pitch to Remain in Ground	
Accelerate to V _X	
200' AGL	
200' AGL	
300' AGL	
300' AGL	Flaps 0°
300' AGL	Flaps 25°
300' AGL Short-Field Takeoff Before Taking Runway Use maximum available runway, li	Flaps 0°Flaps 25° ne up on centerline:
300' AGL Short-Field Takeoff Before Taking Runway. Use maximum available runway, li Brakes.	Flaps 0 ^c Flaps 25 ^c Flaps 0 ^c Flaps 0 ^c Flaps 0 ^c Flaps 0 ^c Flaps 25 ^c Flaps 25 ^c Flaps 25 ^c
Short-Field Takeoff Before Taking Runway. Use maximum available runway, li Brakes. Throttle. Full Pov	Flaps 0°Flaps 25° ne up on centerline:Hold wer/ Verify Max. RPM
Short-Field Takeoff Before Taking Runway. Use maximum available runway, li Brakes Throttle Full Pov Engine Gauges	Flaps 0° Flaps 25° ne up on centerline: Hold wer/ Verify Max. RPM Verify in the green
Short-Field Takeoff Before Taking Runway. Use maximum available runway, li Brakes. Throttle. Full Pov Engine Gauges. Brakes.	Flaps 0° Flaps 25° ne up on centerline: Hold wer/ Verify Max. RPM Verify in the green
Short-Field Takeoff Before Taking Runway. Use maximum available runway, li Brakes. Throttle. Engine Gauges. Brakes. "Airspeed Alive"	Flaps 0° Flaps 25° ne up on centerline: Hold wer/ Verify Max. RPM Verify in the green
Short-Field Takeoff Before Taking Runway Use maximum available runway, li Brakes Throttle	Flaps 0°Flaps 25° ne up on centerline:Hold wer/ Verify Max. RPMVerify in the greenRelease
Short-Field Takeoff Before Taking Runway. Use maximum available runway, li Brakes. Throttle	Flaps 0°Flaps 25° ne up on centerline:Hold wer/ Verify Max. RPMVerify in the greenRelease
Short-Field Takeoff Before Taking Runway. Use maximum available runway, li Brakes. Throttle	Flaps 0° Flaps 25° ne up on centerline: Hold wer/ Verify Max. RPM Release
Short-Field Takeoff Before Taking Runway. Use maximum available runway, li Brakes. Throttle	Flaps 0°Flaps 25° ne up on centerline:Hold wer/ Verify Max. RPMVerify in the greenRelease"Rotate"
Short-Field Takeoff Before Taking Runway. Use maximum available runway, li Brakes. Throttle. Full Pov Engine Gauges. Brakes. "Airspeed Alive" Warrior "50 Knots" Archer "55 Knots" Accelerate	Flaps 0° Flaps 25° ne up on centerline: Hold wer/ Verify Max. RPM Release "Rotate" "Rotate"
Short-Field Takeoff Before Taking Runway. Use maximum available runway, li Brakes. Throttle	Flaps 0° Flaps 25° ne up on centerline: Hold wer/ Verify Max. RPM Release "Rotate" "Rotate"
Short-Field Takeoff Before Taking Runway. Use maximum available runway, li Brakes. Throttle. Full Pov Engine Gauges. Brakes. "Airspeed Alive" Warrior "50 Knots" Archer "55 Knots" Accelerate	Flaps 0° Flaps 25° ne up on centerline: Hold wer/ Verify Max. RPM Verify in the green Release "Rotate" "Rotate" "Vx Flaps 10°
Short-Field Takeoff Before Taking Runway. Use maximum available runway, li Brakes. Throttle. Full Pov Engine Gauges. Brakes. "Airspeed Alive" Warrior "50 Knots" Archer "55 Knots" Accelerate	Flaps 0° Flaps 25° ne up on centerline: Hold wer/ Verify Max. RPM Verify in the green Release "Rotate" Vx Flaps 10°

Soft-Field Landing	
Final Approach Speed	70 KIAS
Touchdown softly, while holding the	
3,	
ground as long as possible. Avoid a	

will cause weight to be transferred to the nose wheel. Continue holding the controls full aft as if taxiing on a soft

Short-Field Landing

surface.

Traffic Pattern

Downwind	90 KIAS
Perform Memory Item	BC-GUMPS
Before Landing Checklist	Complete
(Abeam the Landing Point):	
Power	1700 RPM Below
Below 103 KIAS	F1aps 10°
Pitch & Trim for	90 KIAS
Base	F1aps 25°
Pitch & Trim for	80 KIAS
Final	F1aps 40°
Pitch & Trim forFir	ial Approach Speed

Go Around/Missed Appro	oach
Cram	Max Power
Climb	Pitch V _Y Clear
(when landing with use of	
Flaps	259
Positive Rate at Vy	10°
Flaps	0°
Cool	Carb. Heat Off
Call	Go around/Missed Approach

Piper Warrior Flight Maneuvers

Revised: 12/26/2018

Takeoff and Landing Procedures

-

Revised: 12/26/2018

 V_A = 98 - 113 KIAS --- Maneuvering Speed V_G = 76 KIAS --- Best Glide Speed

NORTH STAR AVIATION **APPENDICIES**NORTH STAR AVIATION **APPENDICIES**NORTH STAR AVIATION **APPENDICIES**

Piper Seminole

Normal Takeoff
Flaps0°
Line up on runway centerline:
Brakes Hold
Rwy/Compass/DGVerify aligned ("33,33,33" etc.) Throttles2000 RPM
Engine GaugesVerify in the green
BrakesRelease
Throttles
"Airspeed Alive"
"75 Knots""Rotate"
Positive RateGear Up
PitchVY
TrimAdjust for Vy
After Reaching 500' AGL
Throttles25" MP
Props2500 RPM
Above 1000' AGL and Clear of PatternClimb Check
Short-Field Takeoff
Flaps0°
Use maximum available runway, line up on centerline: BrakesHold

Use maximum available runway, lin	e up on centerline:
Brakes	•
Throttles	2000 RPM
Engine Gauges	
Throttles	
	Verify Max. RPM
Brakes	Release
"Airspeed Alive"	
"70 Knots"	"Rotate"
Positive Rate	Gear Up
Pitch	Vx
Clear of obstacle	Pitch V _Y
Trim	Adjust for Vy
After Reaching 500' AGL	
Throttles	25" MP
Props	2500 RPM
Above 1000' AGL	Climb Check

Short-Field Landing
Final Approach Speed75 KIAS
Slow to 75 KIAS on short final. Keep a constant angle of
descent to the touchdown point while slowing the airplane
to allow for a touchdown with minimal floating. After
touchdown, apply maximum effective braking. If
simulating a short field, announce "Simulated Max.
Braking" and apply normal braking.

Traffic Pattern	
Downwind	100 KIAS
Perform Memory Item	BCC-GUMPS
Before Landing Checklist	Complete Abeam
Landing Point:	
Power	Reduce
Below 111 KIAS	*Flaps 10°
Pitch & Trim for	100 KIAS
Base	*F1aps 25°
Pitch & Trim for	90 KIAS
Final	*Flaps 40°
Pitch & Trim for	Final Approach Speed
500' AGL	Final Gear Check
	(Announce Safe to Land)

Single Engine Traffic Pattern (Simulated)	
Downwind 100 KIAS (88 KIAS if n	eeded)
Before Landing Checklist	Complete
Abeam Landing Point	Gear Down
PowerR	educe/As Required
Pitch & Trim for	100 KIAS
Base	*Flaps 10°
Pitch & Trim for	90 KIAS
Final	*Flaps 25°
Pitch & Trim for	90 KIAS
500' AGL	.Both props fwd. 1
Committed to Land and Runway Made	
Reduce power slowly and flare airplane	e.

Go Around/Missed Approach (All Engines)		
Cram	Mixt	ure, Prop, Throttle Full fwd.
Climb.		Pitch for Vy
Clean		F1aps 25°
	Positive Rate	Gear Up, Flaps 10°
		Flaps 0°
Cool	Cow1 F	laps As Req./Carb. Heat Off
Call		Go around/Missed Approach

Co Around/Missed Annyoach (Single Engine)

GO ATO	unu/Misseu Approach (Single Engine)	
Above (Committed to Land Altitude	
Cram	nMixture/Prop/Throttle - Full Forward	
Climb	Pitch for Vy	
Clean	Flaps 25°	
	Positive Rate Gear Up	
	Flaps 10°	
Cool	Cowl Flaps As Req./Carb. Heat Off	
Call	Go around/Missed Approach	
	••	

^{*}When adding flaps in the Seminole, the pilot will announce and verify:

"Three green, one in the mirror"

Revised: 12/26/2018 Takeoff and Landing Procedures

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North Star

Instrument Flight/Approach Procedures



North Star Aviation Minnesota State University, Mankato

The 5 T's

Will be executed and verbalized each time when initiating a turn; intercepting a course, crossing a fix/NAVAID, etc.

Turn (turning to heading)
Time (start/note/stop as needed)
Twist (hdg. bug, CDI, HSI, etc.)
Throttle (adjust for alt/airspeed)
Talk (radio call/report as required)

WIRE Checklist

Weather	Obtain
Instruments	DG/Altimeter Set
Radios	Set
(Load freq.'s	needed for appch.)
Environment	Approach Brief

Final Approach Fix Checklist

radar contact, you are not required to

C – Cowl Flaps (Seminole)
G – Gas (Fuel Selector)
U – Under Carriage
M – Mixture
P – Prop Controls
S - Seatbelts

make this report. If approaching a nontowered airport, report aircraft position

 Set
 on the CTAF.

 Set
 IAF Memory Checklist

 B - Boost Pumps (fuel)
 C - Carb Heat

Selecting the GPS for Navigation

ociceting the	OI DIOI INNVISATION
Tune	.Enter "Direct To", etc
Select	GPS*
Twist	Course on CDI/HSI
Steer	To intercept course
*Use the "CI	OI" button on the GPS t
select the na	v. source. If using an
aircraft with	a glass display, select th
appropriate source on the PFD.	

Standard Call Outs

These standard call outs help build habits that increase situational awareness. The use of these call outs has been shown to reduce the risk of pilots falling behind the airplane in the IFR environment.

Departure

Taking the Runway for Takeoff

Verbally verify the DG, compass, and runway numbers are aligned. (i.e. The pilot looks at the heading indicator, compass, and runway numbers and says "33, 33, 33", verifying the correct indications.

Airspeed Alive

When the airspeed indicator needle comes off the wall, the pilot will call "Airspeed Alive".

Rotation

Example: "55 knots, rotate"

Climb/Descent

Approaching Assigned Altitude

"1000 To Go" "500 To Go" "100 To Go"

Approach

"Localizer Alive/Final Approach Course Alive"

"Glideslope Alive"

Approach Mode

Before reaching the final approach fix, the pilot will verify that the GPS has sequenced into approach mode. This will happen when within 2nm of the final approach fix.

Approaching MDA/DA

The pilot will make the 1000', 500', and 100' to go calls when approaching the MDA/DA

"Minimums" - The pilot will announce arrival at the MDA/DA.

"Runway in sight"- The pilot will make this call upon seeing one of the visual references listed in 91.175 in sight.

"Missed Approach" - The pilot will announce when initiating a missed approach.

Revised: 12/26/2018

Instrument Flight/Approach Procedures

^{1:} This is done in the event that an actual go-around using both engines becomes necessary while performing single engine approaches/landings for training purposes.

NORTH STAR AVIATION **APPENDICIES**NORTH STAR AVIATION **APPENDICIES**NORTH STAR AVIATION **APPENDICIES**

Piper Seminole Instrument Approach Procedures

Circling Approach to Landing (All Engines)

beginning final descent. After intercepting PAPI/VASI

......Flaps 25°, Pitch 90 KIAS, begin descent Final......Flaps 40°, Pitch 80 KIAS

Straight-In Approach to Landing (Single Engine)

Before IAFWIRE Checklist Complete¹
Final Approach Speed100 KIAS
(88 KIAS if needed for performance)

Before FAF

.....Before Landing Check Complete
At FAF.....Final Approach Fix Checklist¹
Runway Environment in Sight^{2,3}

.....Flaps 25°, Pitch 90 KIAS

Committed to Land and Runway Made

Both props fwd. (when using sim. feather)⁴ Reduce throttle slowly and flare airplane

Circling Approach to Landing (Single Engine)

Single engine circling approaches should be avoided if possible. If necessary to perform a circling approach, enter a normal traffic pattern after the runway environment is in sight and perform the normal procedures for a single engine traffic pattern. This includes the procedures for extension of the gear and flaps. For performance purposes, it may be required to delay extension of the gear until beginning descent below MDA. Extra caution must be taken in this case to ensure the gear is down and locked prior to landing.

Piper Warrior Instrument Approach Procedures

Straight-In Approach to Landing
Before Reaching IAFWIRE Checklist Complete¹
Final Approach Speed90 KIAS
Before Reaching FAFBefore Landing Check Complete
At FAFFAF Check Complete¹
Runway Environment in Sight^{2,3}Flaps 25°, Pitch 80 KIAS
.......Flaps 40°, Pitch 70 KIAS

Circling Approach to Landing Procedures are the same as a straight in approach to landing until......

FAF......Leave Flaps 0° Until Downwind
Runway Environment in Sight
Position for circling approach
Enter a normal traffic pattern if practical
Maintain MDA and intercept PAPI/VASI before
beginning final descent.

After intercepting PAPI/VASI

Final.....Flaps 25°, Pitch 80 KIAS, begin descent Final.....Flaps 40°, Pitch 70 KIAS

FAR 91.175 Descent Below MDA or DA/DH

Three things required for descent below MDA/DA.

- Normal maneuvers to landing and a normal rate of descent to the intended runway.³
- Flight visibility not less than that listed in the approach procedure.
- 3. Runway environment in sight, including one of the following references:
 - ---Runway end identifier lights (REILs)
 - ---Visual approach slope indicator
 - ---Runway/Runway markings and lights
 - ---Threshold/Threshold markings and lights
 - --- Touchdown zone/TDZ markings and lights
 - ---Approach Lights
 - Allows descent to 100 ft. above TDZE
 - -Descent below 100 ft. allowed if red terminating bars or red side row bars are in sight.

¹See FAF Fix/WIRE Checklist on previous page. For approaches in the Seminole, the landing gear will be extended at the FAF.

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Revised: 12/26/2018 Instrument Flight/Approach Procedures

Appendix C Contact Information

North Star Aviation, Inc.

North Star Aviation, Inc. 3030 Airport Road North Mankato, MN 56001

Phone Directory.....(507)625-6006

Fax.....(507)625-6130

Web.....www.flymankato.com

Flight School

Chief Flight Instructor

Christopher Plasek

Office.....(507)625-6006 ext. # 343 Cell(218)251-0205

Email.....cplasek@flymankato.com

Assistant Chief Flight Instructors

Jeff Peterson

Cell(952)374-8187

Email....jpeterson@flymankato.com

Ethan Plunkett

Cell(218)780-7842

Email.....eplunkett@flymankato.com

Chief Dispatcher

Cody Howe

Cell(507)621-0440

Email.....dispatch@flymankato.com

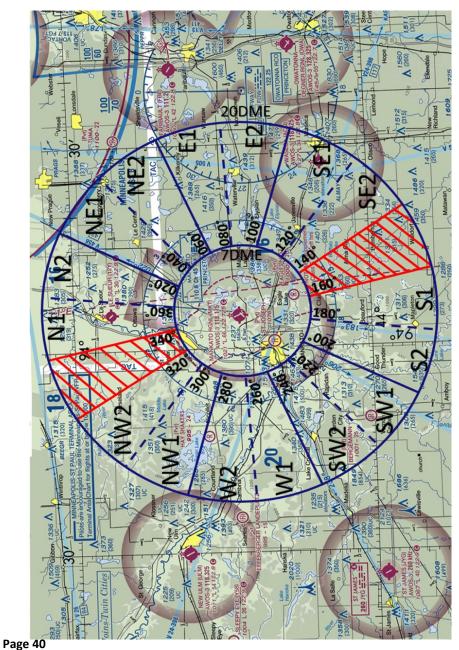
²When flying a precision approach, stay on the electronic glide path until it is necessary to descend below.

³When flying a non – precision approach, maintain MDA until intercepting the PAPI/VASI to ensure normal maneuvers to landing.

⁴This is done in the event that an actual go-around using both engines becomes necessary while performing single engine approaches for training purposes.

NORTH STAR AVIATION **APPENDICIES**NORTH STAR AVIATION **APPENDICIES**

Appendix D Practice Areas



Appendix E Briefings

Passenger

Seatbelt Use
Opening and Closing Door
Emergency Exits
Emergency Equipment (ELT, Fire Extinguisher)
Help Watch for Traffic
Sterile Cockpit Taxi, Takeoff, and Landing

<u>Taxi</u>

Runway in Use
Runway Length Available/Required
Route to Runway and Runup
Location of Hot Spots
Who Will Be PIC
Positive Exchange of Flight Controls

Takeoff Briefing (PA-28)

-Type of Takeoff (Normal, Soft or Short)
-Rotate____ Climb Out____
-Problem on Takeoff Roll – Abort
-Problem After Take Off Below 500' AGL—Land
Straight Ahead and Obstacles to Consider are

-Problem After Take Off Below 1000' Turn No More than 90 $\,$

-Problem After Take Off 1000' and Above—Land Where Practical

Takeoff Briefing (PA-44)

-Type of Takeoff (Normal or Short)

-Rotate____ Climb Out___

-Problem on Takeoff Roll - Abort

-Engine Failure after Takeoff – Gear Extended Maintain V_{YSE} – 88 kts Land Straight Ahead

-Engine Failure After Takeoff – Gear Retracted

Maintain V_{YSE} – 88 kts Attempt return for Landing

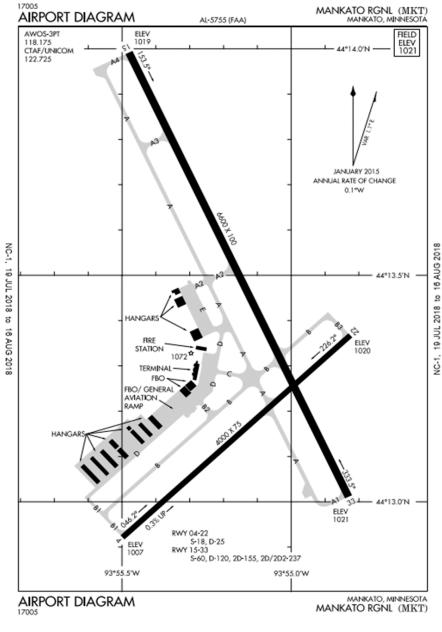
Runway_____

Committed to Land Altitude Today

Instrument Takeoff Briefing

Initial Heading/Altitude Airspeed Limits if Any Departure Procedures Emergency Return Plan NORTH STAR AVIATION **APPENDICIES**NORTH STAR AVIATION **APPENDICIES**

Appendix F Airport Diagram



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Appendix G Misc. Charts

PIVITOL ALTITUDE CHART (GS^2/11.3)	
GROUNDSPEED	PIVOTAL ALTITUDE
70	434
80	566
90	717
100	885
110	1071
120	1274
130	1496